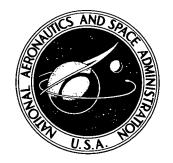
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HEAT-TRANSFER AND PRESSURE DISTRIBUTIONS FOR LAMINAR SEPARATED FLOWS DOWNSTREAM

OF REARWARD-FACING STEPS WITH AND WITHOUT MASS SUCTION

by Ronald D. Brown and Antoni K. Jakubowski AERONUTRONIC LIBRARY

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16. Abstract

Heat-transfer and pressure distributions were measured for laminar separated flows downstream of rearward-facing steps with and without mass suction. The flow conditions were such that the boundary-layer thickness was comparable to or larger than the step height. For both suction and no-suction cases, an increase in the step height resulted in a sharp decrease in the initial heat-transfer rates behind the step. Downstream, however, the heat transfer gradually recovered back to less than or near attached-flow values. Mass suction from the step base area increased the local heat-transfer rates; however, this effect was relatively weak for the laminar flows considered. Even removal of the entire approaching boundary layer raised the post-step heat-transfer rates only about 10 percent above the flat-plate values.

Post-step pressure distributions were found to depend on the entrainment conditions at separation. In the case of the solid-faced step, a sharp pressure drop behind the step was followed by a very short plateau and a relatively fast recompression. For the slotted-step connected to a large plenum but without suction, the pressure drop at the base was much smaller and the downstream recompression more gradual than that for the solid-faced step.

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SUMMARY

Heat-transfer and pressure distributions were measured for laminar separated flows downstream of rearward-facing steps with and without mass suction. The flow conditions were such that the boundary-layer thickness was comparable to or larger than the step height. For both suction and no-suction cases, an increase in the step height resulted in a sharp decrease in the initial heat-transfer rates behind the step. Downstream, however, the heat transfer gradually recovered back to less than or near attached-flow values. Mass suction from the step base area increased the local heat-transfer rates; however, this effect was relatively weak for the laminar flows considered. Even removal of the entire approaching boundary layer raised the post-step heat-transfer rates only about 10 percent above the flat-plate values.

Post-step pressure distributions were found to depend on the entrainment conditions at separation. In the case of the solid-faced step, a sharp pressure drop behind the step was followed by a very short plateau and a relatively fast recompression. For the slotted-step connected to a large plenum but without suction, the pressure drop at the base was much smaller and the downstream recompression more gradual than that for the solid-faced step.

INTRODUCTION

The study of the high-speed flow field past a rearward-facing step is of considerable interest in the aerothermodynamic design of various flight configurations, including large lifting entry vehicles. In particular, the possible use of overlapping sliding metallic heat-shield joints calls for a study of the separated region behind a step. The conditions of particular interest are characterized by a thick boundary layer (i.e., the step height h is comparable to or less than the boundary-layer thickness δ) and mass suction from the separated area. The studies of various aspects of a rearward-facing step or wedge are

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numerous, but the vast majority of published works have centered on the case where the boundary-layer thickness is smaller than the step height. Examples of such investigations are Rom and Seginer (ref. 1), Holloway, Sterrett, and Creekmore (ref. 2), Weiss and Weinbaum (ref. 3), Scherberg and Smith (refs. 4, 5), Hama (ref. 6), Adamson (ref. 7), Batt and Kubota (ref. 8), and more recent works by Erdos and Zakkay (ref. 9) and Wu and Su (ref. 10). Experimental information on the case of $h/\delta \le 1$ is almost nonexistent and, to our knowledge, no experiments have been made that include mass suction from the separated area.

The purpose of the present investigation was to determine experimentally the heat-transfer and pressure distributions for laminar separated flows downstream of a rearward-facing step with and without mass suction under conditions where the boundary-layer thickness is of the same order as the step height. The program was a joint effort between the Virginia Polytechnic Institute and State University (under an NASA grant) and the NASA Langley Research Center. The experiments were conducted in an arc-heated wind tunnel of the Langley entry structures facility. Apparatus, experimental techniques, and test conditions are described, and data collected during this study are presented, along with a discussion of the basic results.

SYMBOLS

width of model b enthalpy -1 step height h L length of surface ahead of step Mach number M m mass flow rate $N_{\mathbf{Re}}$ Reynolds number Reynolds number based on h, $u_{\infty}h/\nu_{\infty}$ $^{N}\mathrm{Re}_{\infty,h}$ Reynolds number based on L, $u_{\infty}L/\nu_{\infty}$ $^{N}_{Re_{\infty,L}}$

 $N_{\text{Re}_{\infty}}$ Reynolds number based on x, $u_{\infty}x/\nu_{\infty}$

 N_{St} Stanton number, $\dot{q}/\rho_{\infty}u_{\infty}H$

p pressure

q heat-transfer rate

R gas constant

T absolute temperature

u velocity

w nondimensional mass suction rate, $\dot{m}_s/\rho_{\infty}u_{\infty}bh$

x distance downstream of leading edge

Δx distance downstream of step

x_e distance downstream of nozzle exit

y transverse distance measured from center line; distance from surface

δ boundary-layer thickness

δ* boundary-layer displacement thickness

v kinematic viscosity

ρ density

Subscripts:

BL boundary layer

max maximum value

0 stagnation condition (see fig. 1)

ref reference condition

s suction

st static

step step location on flat plate

t total, pitot (see fig. 1)

ts test section

w wall condition

∞ free-stream condition

APPARATUS

Figure 1 presents a simplified diagram of the experimental setup used in this investigation. The arc-heated wind tunnel consists of a magnetically stabilized ac arc heater, a plenum chamber, a 15° conical supersonic nozzle, a test section, and a diffuser which exhausts into a vacuum sphere at a minimum pressure of about 1 mm Hg. A multistage steam ejector provides sufficient vacuum for the tunnel operation. The wind tunnel is a free-jet type and has a run duration of about 10 minutes to over 1 hour, depending on the mass flow rate. The test medium was arc-heated air. The water-cooled supersonic nozzle had a 7 cm (2.75 in.) diameter throat and a 22.9 cm (9 in.) diameter exit section. The nominal stagnation pressure and enthalpy were varied from 0.18 to 1.96 atm and from 3.5 MJ/kg to 12.8 MJ/kg (1500 to 5500 Btu/lbm), respectively. The models were mounted on a water-cooled retractable strut.

Models and Instrumentation

Two basic models were used in this investigation. Model I (fig. 2) was designed for tests without suction while model II (fig. 3) was designed and used primarily for tests with suction. Each model was a wedge designed to provide two-dimensional flow on the test surface and had a sharp leading edge, a rearward-facing step, and adjustable test surface heights. Each model was constructed of water-cooled copper.

Model I (fig. 2) had a flat surface preceding the step that was 12.5 cm (4.91 in.) long. This surface was instrumented with a pressure orifice that was located 3.1 cm (1.22 in.) upstream of the step. Another pressure orifice was located in the center of the

rearward-facing step (0.51 cm from the top). The 11.4 cm (4.5 in.) test surface down-stream of the step was provided with one of two interchangeable reattachment plates. The heat-transfer plate was 0.74 mm (0.029 in.) thick stainless steel that was instrumented down the center line by spot welding 17 No. 30 gage chromel-alumel thermocouples to the underside. The pressure-distribution plate was 0.635 cm (0.25 in.) thick water-cooled copper and was instrumented by silver-soldering to the pressure orifices 14 stainless-steel tubes 1.2 mm (0.047 in.) inside diameter that were 2.4 m (8 ft) in length. The step height was varied by changing the thickness of the shims beneath the reattachment plates.

Model II (fig. 3) had a flat surface preceding the step that was 10 cm (3.925 in.) long. There was no pressure orifice in this surface because nearly the entire area of the rearward-facing step was used as a suction slot. The heat-transfer plate was identical to that of model I, but the pressure plate was restricted to 12 pressure orifices located immediately downstream of the step.

The heat-transfer data were recorded on a central digital data system at 0.025 sec intervals. The pressure-distribution data were also recorded on the central digital data system but at intervals of approximately 1.0 sec.

Mass-Suction Installation

The experimental test setup used for the suction experiments on model II is shown schematically in figure 1. The hot air removed from the separated region behind the step was passed through the water-cooled strut, through two 7.6 cm (3 in.) diameter flexible hoses that were connected to the two suction outlets of the model, a vacuum pump, an instrumented section to measure the mass flow rate, and then reintroduced into the test section. The vacuum pump was a mechanical blower with a pumping capacity of 1620 liters/sec (3435 standard ft³/min). The instrumented section was connected to the discharge side of the vacuum pump and equipped with static- and total-pressure sensors and stagnation- and wall-temperature sensors to provide information necessary for evaluation of suction mass flow rates. Pressures were measured by means of strain-gage pressure transducers, and total and wall temperatures were sensed by thermocouple probes. The pressure and temperature data were recorded by a central digital data system. Pressure taps in the suction duct upstream of the vacuum pump provided information to determine that the flow conditions in that section had been stabilized.

The vacuum pump available for these experiments operated at constant speed, and, consequently, there was no convenient way to vary mass suction flow rates at a given test condition. Only small variations of the suction flow rate were obtained by blanking off one of the exhaust ports on the model. The vacuum pump failed before the planned program was completed, and information concerning effects of the mass suction is therefore limited.

PROCEDURES

Flow Conditions

The models were mounted parallel to the supporting strut to minimize blockage and reduce the spillage of the test gas about the diffuser inlet scoop. This arrangement placed the attachment plates approximately parallel to the tunnel viewing ports and precluded any effective use of optical means of flow observation. A nominal zero angle of attack was used throughout the tests.

The stagnation test conditions were varied by changing the tunnel mass flow rate (from 0.045 kg/sec to 0.45 kg/sec) and the power input to the arc heater. Prior to model insertion, the pitot pressure and the stagnation-point heat-transfer rate near the model leading-edge test position were determined by using standard retractable devices (impact pressure probe and hemispherical calorimeter) injected sequentially into the test stream. These measurements, combined with that of the pressure in the plenum chamber, served to establish stagnation and free-stream conditions for each individual run. In addition, during each experiment, several electrical and gas dynamic measurements were performed to provide or supplement information on the conditions of the experiment. These measurements included the arc current, arc voltage, arc power, tunnel-air mass flow rate, test-chamber pressure, temperature of the instrumented plate being tested, and energy balance data on the arc heater. The stagnation enthalpy was calculated from stagnation-point heat-transfer and pressure measurements. The theory of Fay and Riddell (ref. 11) was used. A unit Lewis number and thermodynamic equilibrium at the surface were assumed, and the viscosity from Hansen (ref. 12) was used. The calculated enthalpy and the measured total pressure p_0 (plenum chamber pressure) were used to define the total temperature (plenum) from a Mollier diagram.

Free-stream conditions at any location in the test stream were calculated, based on a frozen isentropic expansion from the total temperature and pressure (plenum) to the local condition defined from a pitot pressure survey. Flat-plate boundary-layer parameters were determined by using frozen flow edge conditions and a finite-rate boundary-layer calculation with an equilibrium catalytic wall. A detailed description of the computation is given in reference 13. Heat-transfer and pressure measurements for all test conditions are listed in tables I to VI. The nominal range of test and model conditions is as follows:

| Stagnation pressure, p ₀ , atm | 0.18 to 1.96 |
|---|--------------|
| Stagnation enthalpy, H ₀ , MJ/kg | 3.5 to 12.8 |
| Stagnation temperature, T ₀ , K | 2700 to 5500 |
| Mach number, M _∞ | 3.95 to 4.27 |

| Reynolds number per unit length, N _{Re} per cm |
|--|
| Reynolds number based on prestep surface length, $N_{{\rm Re}_{\infty, L}}$ 2.9×10^3 to 2.9×10^4 |
| Reynolds number based on step height, $N_{\text{Re}_{\infty,h}}$ 0 to 2200 |
| Ratio of wall temperature to free-stream stagnation temperature, |
| T_{w}/T_{0} 0.055 to 0.11 |
| Step height, h, cm |
| Ratio of step height to boundary-layer thickness, h/δ 0.1 to 2.4 |
| Dimensionless mass suction rate, w |

Flow Uniformity

Pitot pressure surveys along the center line of the test stream revealed the presence of an axial pressure gradient in the jet (underexpansion) as shown in figure 4. The presence of the pressure gradient was accounted for in establishing free-stream conditions and boundary-layer parameters along the surface of the models.

To examine transverse uniformity or two dimensionality of the flow along the instrumented sections of the models, transverse surveys of pressure distribution were made at a few stations along the instrumented section. The results of this examination, shown in figures 5(a) to 5(b) for h=0 cm and h=0.97 cm, indicate that the flow near the center line was essentially two dimensional over a distance of at least 6.5 cm (2.5 in.). In view of the longitudinal pressure gradient in the jet and the limited width of the models, great caution must be exercised when interpreting data collected far downstream of the step $(\Delta x > 6.5$ cm (2.5 in.)). In addition, under some flow conditions, the rear part of the model was influenced by back pressure effects from a shock-wave system generated by the supporting strut. Since the main object of this investigation was to study the separated flow region immediately downstream of the step, the flow structure downstream of the reattachment was of secondary interest.

Mass-Suction Experiments

In the experiments involving mass suction (model II), the vacuum pump in the suction circuit was started before the model was injected into the jet. For most of the tests, less than 0.8 sec was needed to stabilize pressures in the suction circuit after the model had been injected.

The suction mass flow rate \dot{m}_{s} was calculated from the mass-balance equation

$$\dot{m}_{s} = A \sqrt{\left(2p_{st}/RT_{0}\right)\left(p_{t} - p_{st}\right)}$$
(1)

where A is the flow area of the instrumented section, and T_0 , p_{st} , and p_t are total temperature, static pressure, and average total pressure in that section. The absolute value of $(p_t - p_{st})$ was very small (on the order of 0.1 mm Hg for some tests) and difficult to measure. In addition, the available pressure measuring system had a slow response time and experienced some zero drift when cycling to atmospheric pressure for model changes. To minimize these problems, a calibration procedure was applied before and after each series of runs. The procedure consisted of checking zero intercepts and slopes near zero pressure of the transducers involved in measuring p_t - p_{st} and comparing the outputs against a calibrated pressure gage.

Heat-Transfer Tests

Heat-transfer data were obtained from measurements of transient skin temperatures resulting from a stepwise increase in stagnation temperature. The model, initially at room temperature, was suddenly (in less than 0.25 sec) exposed to the hot air flow where it remained from 1 to 3 sec. Skin temperatures were recorded at a rate of 40 measurements per sec. Heat-transfer data were computed by using the simplified heat-balance equation

$$\dot{q} = c_w \rho_w d_w \frac{dT_w}{dt}$$
 (2)

where c_W is the specific heat of the wall material, ρ_W is the density of the wall material, d_W is the wall thickness, T_W is the wall temperature, and t is time. Equation (2) presupposes a constant temperature through the model skin, negligible lateral heat flow, negligible heat flow to the interior of the model, and no heat losses through radiation. The experiments with the mass suction from the separated area involved a time delay required to stabilize pressures in the suction circuit. Consequently, the heat-transfer measurements taken after the flow conditions were stabilized might be affected by the lateral heat conduction in the plate. To examine this possibility, plots of skin temperature against time were checked for several test conditions. The results showed that the slope dT_W/dt remained very nearly constant for at least 3 seconds, that is, for a period longer than that required to stabilize the flow in the suction duct (less than 0.8 sec). Therefore, it was concluded that the heat conduction did not have any significant effect on the heat-transfer measurements. The overall error of the heat-transfer measurements was estimated to be about 10 percent.

Pressure Tests

The strain-gage pressure transducers were mounted outside the test section for protection from the hot gases in the tunnel and were connected to the pressure taps on the

model by 1.2 mm (0.047 in.) i.d. tubing that was 2.4 m (8 ft) in length. Thus, the response time of the pressure measuring system was slow, ranging from 20 sec to about 2 min. Consequently, the model with pressure-instrumented reattachment plate had to be kept in the jet for relatively long periods, and the plate was water cooled.

RESULTS AND DISCUSSION

Heat-Transfer Distribution on Model I (No Suction)

The heat-transfer data obtained at all model stations and test points are presented in table V. The stagnation and free-stream conditions for each test can be determined from table I by using the run number.

The classical laminar flat-plate heat-transfer correlation of Stanton number and Reynolds number is shown in figure 6 with experimental heat-transfer data from model I. The correlation between the experiment and classical theory is good over the Reynolds number range of 3×10^3 to 3×10^4 . Therefore, the undisturbed flow in these tests is clearly laminar.

The typical effect of step height on the heat-transfer distribution along the center line of the reattachment plate is shown in figure 7. For the different step heights, free-stream conditions were similar but not necessarily identical. The dashed lines indicate data that may be influenced by side effects or disturbance in the external flow. The figure shows that an increase in step height produces a decrease in heat transfer just behind the step. In all cases this drop is followed by a gradual increase in heating to a value less than or nearly equal to that for attached flow. The maximum heat-transfer rate occurs at a distance from the step that varies from 6 step heights for the largest step (1.02 cm) to about 14 step heights for the smallest step (0.16 cm). The same general trends shown in figure 7 held for the other test results contained in table V.

These tests demonstrate that for a small rearward-facing step in laminar flow without suction no significant increase in heating occurs downstream of the step. This indicates that metallic heat-shield systems which would have small steps on the order of typical material thickness will not cause excess heating downstream of the joints between panels, provided the joint is sealed against air inflow.

The range of conditions where peak heating downstream of the step is less than heating on the plate approaching the step can be inferred from figure 8. This figure includes data obtained by Rom and Seginer (ref. 1), Smith (ref. 5), and several points evaluated on the basis of measurements reported by Holloway, Sterrett, and Creekmore (ref. 2), as well as data from the present investigation. It should be noted that the abscissa on figure 8 $\left(N_{Re_{\infty,h}}\right)$ is proportional to the square of the ratio of step height to

boundary-layer thickness. This follows because, for a given geometry L/h and free-stream Mach number M_{∞} , the Reynolds number $N_{\text{Re}_{\infty,h}}$ varies approximately as $\left(\frac{h}{\delta}\right)^2$ (in laminar flow, $N_{\text{Re}_{\infty,h}} \approx \frac{L}{h} \left(\frac{h}{\delta}\right)^2$).

Pressure-Distribution on Model I (No Suction)

The wall pressure distributions normalized by the reference pressure measured upstream of the step (p_{ref}) are displayed in figures 9(a) to 9(g). The pressure at the step location ($\Delta x = 0$) was measured on the perpendicular surface of the step. This pressure was not measured for the small step (h = 0.2 cm) because the pressure port was below the pressure plate. The pressure distribution indicates a small region of low pressure along the surface immediately downstream of the step. The length of this region increases with step height h and with free stream Reynolds number $N_{Re_{\infty},L}$. Downstream of the low pressure "plateau" the pressure increases in a few step heights to approximately the attached flow value of the flat plate.

Heat-Transfer Distribution on Model II (Suction)

Heat-transfer data obtained from model II on the 1.02 cm and 0.6 cm steps are tabulated in table VI, and representative points are shown in figures 10(a) to 10(j) and figures 11(a) and 11(d), respectively. The stagnation and free-stream conditions for each test are listed by run number in table II. The suction and no-suction results obtained with model II are compared with heat-transfer distributions for a flat-plate configuration under similar free-stream conditions. The mass suction rate w is defined as the ratio of the average mass flux through the slot \dot{m}_s/bh to a free-stream mass flux. The mass suction rates w varied between 0.2 and 0.5 for the 1.02 cm step and between 0.1 and 0.8 for the 0.6 cm step. In terms of the entire mass flow rate throughout the boundary layer

 $\left(\dot{m}_{BL} = \int_{0}^{\delta} \rho u \; dy\right)$, the mass flow rate \dot{m}_{s} ranged from 40 percent to 120 percent of \dot{m}_{BL} for the 1.02 cm step and 0.6 cm step, respectively. The no-suction experiments were made with the suction duct blanked off at the flexible hose coupling, thus leaving a large internal cavity connected to the suction slot. As a result, the entrainment conditions at separation in the no-suction tests on model II were different from those in the no-suction tests on model I.

Figure 10(a) shows that the effects of suction on the heat transfer downstream of a 1.02 cm step is negligible for w = 0.25 and $N_{Re_{\infty}}$ per cm = 250. From figure 10(a) to figure 10(b) the Reynolds number is doubled for approximately the same relative suction and again the increased heating is negligible. From figures 10(b) to 10(d) stagnation

enthalpy is increased with the same relative suction and Reynolds number. Figures 10(b), 10(c), and 10(d) show a small increase in heat transfer just downstream of the step and the effect diminishes in about 3 or 4 step heights. From figures 10(c) to 10(e) the suction is increased with the Reynolds number and the enthalpy constant. The relative increase in heating is the largest just downstream of the step and approaches the flat-plate heat-transfer value at about 8 step heights. From figures 10(e) to 10(f) the Reynolds number is doubled at the same suction rate, and again the relative increase in heating is the largest just downstream of the step but the heat transfer does not recover to the flat-plate value.

A comparison of figures 10(g) and 10(h) indicates that the repeatability is good at a relative suction rate of $w \approx 0.34$. In figure 10(i), a comparison of two suction rates (w = 0.36 and 0.41) indicates that the increase in suction gives an increase in heating just downstream of the step. The lower suction value was obtained by blanking off one of the exhaust ports of the model, as previously discussed. In figure 10(j), the two relative suction rates were calculated to be equal, but the lower heat-transfer points were obtained with one port on the model blanked off. Thus, it must be concluded that the suction flow measurement is only an approximation.

The heat-transfer data presented in figure 10 indicate that suction causes a large percentage increase in the heat transfer just downstream of the step. In addition, for a constant step height and relative suction rate the location downstream of the step where the heating approaches the flat-plate value appears to be a function of the Reynolds number.

Figure 11(a) shows that a small relative suction rate (w = 0.12) does affect the relative heat transfer just downstream of the 0.6 cm step but the effect diminishes in about 3 or 4 step heights. In figure 11(b), an increase in the relative suction (w = 0.61) carries downstream to approximately 10 step heights where the heat transfer approaches the flat-plate value. The lower relative suction value (w = 0.59) was obtained with the one exhaust port blanked, as previously discussed. It appears that the sonic point was in the duct in the strut so that a factor of 2 in the exhaust area made only a slight difference in the measured suction mass flow rate (they were equal when the measurement accuracy is considered). From figures 11(b) to 11(c), the enthalpy is increased without any effect on the relative heat-transfer distribution or on the location of the point where the heat transfer approaches the flat-plate value. From figures 11(b) to 11(d) the Reynolds number was doubled, and at the highest relative suction value (w = 0.79) the heat transfer reached the flat-plate value at approximately 7 to 8 step heights and exceeded the flat-plate value downstream of this location.

Additional heat-transfer data with suction in table VI show the same general trends. Mass suction through the slotted step increases the local heat transfer. The relative

increase, referenced to the no-suction case, depends on (1) the step height, (2) the mass-suction rate, and (3) the location behind the step. The relative increase is most significant a few step heights immediately behind the step. Location of the maximum heating rate varied between 4 and 6 step heights behind the 1.02 cm step and between 7 and 9 step heights behind the 0.6 cm step. In terms of the maximum heating rates, even large suction rates applied in this investigation did not produce any spectacular peaks, and the effect of mass suction seems to be weaker than might be expected. Nevertheless, at the largest value of w applied in this investigation (w = 0.8 in run 190-5-24, see fig. 11(d)), local heat-transfer rates exceeded the flat-plate values over a significant length of the reattachment plate.

Pressure Distribution on Model II (Suction)

Pressure distributions downstream of the 1.02 cm slotted step with and without mass suction are presented in figures 12(a) to 12(j). The results are compared with those for the flat-plate configuration. Pressure data with mass suction were obtained only on the 1.02 cm step because of the failure of the vacuum pump.

As can be seen in figure 12(a), the low total pressure (small Reynolds number) suction data practically coincided with the no-suction data. An increase in total pressure (approximately $1\frac{1}{2}$ from fig. 12(a) to fig. 12(b)) gives a corresponding increase in flat-plate pressure, but the step suction and no-suction results coincide at approximately the absolute pressure level of figure 12(a) immediately behind the step. The pressure then gradually increases to the flat-plate value at approximately 7 step heights. The pressure exceeds the flat-plate pressure data beyond 7 step heights. The increase in pressure downstream of the step is more gradual for the slotted step with no suction than for the solid step (see fig. 9(b)). This may be explained in terms of the large entrainment volume because the suction exhaust ports were capped.

From figures 12(b) to 12(d), the total enthalpy was increased with only small changes in the Reynolds number and total pressure, and again the suction and no suction data coincide. From figures 12(b) to 12(e), the total pressure, thus Reynolds number, is increased by a factor of about 2 at the same enthalpy, and the suction data begin to deviate from the no-suction data. The mass suction moves the point that the pressure recovers to the flat-plate value upstream approximately 1.5 to 2.0 step heights.

For figures 12(e) to 12(g), the total enthalpy was increased with small changes in Reynolds number and total pressure, with the same type of result that was obtained at the lower enthalpy. Comparison of figure 12(f) with figure 12(h) shows that as the total pressure and Reynolds number are increased by about 2, the effects of the suction become quite pronounced. At about 3 step heights downstream of the step, the pressure with

suction exceeds the pressure with no suction and approaches the flat-plate value at approximately 7 step heights. From figures 12(h) to 12(i), an increase in enthalpy gives the same general results.

It should be noted that the absolute pressure measured at the first pressure port downstream of the step with suction was approximately the same regardless of total pressure or Reynolds number level (varied over approximately a factor of 8).

Figure 12 shows the effects of suction on the pressure distribution downstream of the 1.02 cm step. The suction distorts the pressure distribution so that the pressure is less than the no-suction pressure just downstream of the step but approaches the no-suction pressure at approximately 3 to 4 step heights and remains above the no-suction pressure for the remainder of the instrumented section of the plate.

CONCLUSIONS

Measurements of heat-transfer and pressure distributions for laminar separated flow downstream of a step with and without suction demonstrate the following effects:

- 1. For both suction and no-suction cases, an increase in the step height results in a sharp drop in the initial heating rates behind the step. Downstream of the step, the heat-transfer rates gradually recover back to values less than or near attached flow values. The height of the step controls the heat-transfer rates at the step base and dominates the effects of stagnation enthalpy and pressure.
- 2. Mass suction from the step base area increases the local heating rates, the relative increase being most significant immediately behind the step. In general, however, the effect of mass suction on heat transfer in laminar flow can be termed as weak, and removal of the entire boundary layer raised the post-step heat-transfer rates only about 10 percent above the flat-plate values.
- 3. Pressure distribution downstream of the step was found to depend on the entrainment conditions at separation. In the no-suction solid-step case, pressure dropped sharply immediately behind the step to a level strongly dependent on the step height. This drop was followed by a short pressure plateau and then by a recompression in a short distance. The recompression without suction on the slotted step required a longer distance than with a corresponding solid-step configuration. With mass suction applied, the pressure level at the step base was roughly the same for all Reynolds numbers and suction rates in these experiments.

Langley Research Center,

National Aeronautics and Space Administration,

Hampton, Va., April 17, 1974.

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TABLE I.- TEST CONDITIONS FOR HEAT-TRANSFER MEASUREMENTS ON MODEL I

| $\begin{array}{cccccccccccccccccccccccccccccccccccc$ | | | | | | | | | | | | | | | | |
|--|--------|---------------|-----------------|------------------------|---------------------------|--------|----------------------------------|---|------------------|-------------|------------|---|------|------|------------|-----|
| 0.0 0.0103 0.178 3.95 2900 0.91 0.39 790 2279 4.01 3.11 0.0 0.1129 0.47 0.99 0.911 0.99 0.911 0.99 0.99 0.99 0. | | h, cm | $T_{\rm w}/T_0$ | ^p 0, atm | H ₀ , MJ/kg | T K | p _∞ , atm × 103 | $\begin{array}{c} \rho_{\infty}, \\ \mathrm{kg/m}^3 \\ \times 10^3 \end{array}$ | H _® , | u, m/sec | ¥ 8 | $^{ m N}_{ m Re}{}_{\infty, L}$ $	imes 10^{-3}$ | | | o, cm | 9/u |
| Color Colo | , | d | | | ć | 6 | 6 | 6 | 90, | 2 | 5 | | c | | | |
| | 03-05 | • • | | 0.205 | 0 • 4 0 6 • 4 0 | 3400 | 1.001 | 0.37 | 854 | 2531 | 60.4 | 3.11 | 000 | 1.29 | | |
| 1.0 0.00 0.0.073 0.00 | 03-00 | 0 | | 0.221 | 6.97 | 3 900 | 1.03 | 0.32 | 897 | 2770 | 4.22 | 2.83 | 0 | 1,33 | | |
| 1.00 | 03-12 | 0.0 | 0 | 0.233 | 7.90 | 4100 | 0.96 | 0.30 | 922 | 2855 | 4.27 | 2,75 | 0.0 | 1.36 | 10 | • |
| 1.00 | 03-15 | 0.0 | .36 | 0.253 | 10.23 | 4660 | 1.16 | 0.31 | 1011 | 3051 | 4.19 | 2.73 | 0.0 | 1.40 | 10 | |
| -2.1 0.0 0.055 0.271 12.79 5500 1.62 0.284 04.29 5.28 0.0 0.0 0.99 0.37 0.00 0.00 0.00 0.00 0.00 0.00 0.00 | 03-18 | 0.0 | • | 0.265 | 11.62 | 5200 | 1.17 | 0.29 | 1150 | 3260 | 4.25 | 2.53 | 0.0 | 1.48 | 10 | • |
| | 03-21 | 0.0 | • | 0.277 | 12.79 | 5500 | 1.22 | 0.28 | 1168 | 3389 | 4.29 | 2.52 | 0.0 | 1.43 | | • |
| 9-27 9.0 0.0087 0.380 6.581 3430 1.84 0.67 874 2535 4.09 5.54 0.0 0.0 0.99 0.37 0.00 9-38 0.0 0.0081 0.440 7.90 4.200 1.84 0.60 979 2876 4.15 5.41 0.0 0 0.0 0.99 0.37 0.00 9-39 0.0 0.0011 0.440 7.90 4.200 1.89 0.60 979 2876 4.15 5.41 0.0 0 0.0 0.001 0.39 0.001 9-39 0.0 0.0011 0.4495 10.44 4950 2.14 0.59 1158 3240 4.22 5.05 0.0 0.0 0.099 0.37 0.00 9-30 0.000 0.0045 0.0485 3.49 3160 3.34 1.38 8792 2.24 4.02 10.89 0.0 0.0 0.099 0.37 0.00 9-30 0.000 0.0045 0.0485 3.49 3180 3.34 1.38 862 2.392 4.02 10.87 0.0 0.00 0.005 0. | 03-24 | 0.0 | • | 0.360 | 4.88 | 3230 | 1.62 | 0.04 | 831 | 2445 | 4.12 | 5.28 | 0.0 | 66.0 | ~ | • |
| 1.00 | 03-27 | o•c | • | 0.380 | 5.81 | 3430 | 1.84 | 0.67 | 874 | 2535 | 60° | 5.54 | 0.0 | 0.98 | ~ | ٠ |
| 0.0 0.006 0.0440 0.440 0.450 0.207 0.508 0.00 4.21 5.05 0.00 1.03 0.420 0.00 0.00 0.006 0.4450 0.207 0.508 0.207 0.508 0.00 1.03 0.440 0.00 0.495 0.207 0.508 0.207 0.208 0.207 0.008 0.0095 0.0495 0.207 0.208 0.207 0.208 0.00 0.0095 0.0495 0.207 0.208 0.207 0.208 0.208 0.0095 0.0495 0.208 0.207 0.208 | 03-30 | o • • | • | 0.400 | 6.51 | 3700 | 1.84 | 0.63 | 897 | 2667 | 4.15 | 5,41 | 0.0 | 0.99 | ~ | |
| 1.00 0.00 | 03-33 | • • | • | 0.440 | 7.90 | 4200 | 1.99 | 0.00 | 616 | 2876 | 4.19 | 5.25 | 0.0 | 1.01 | ~ | • |
| 1.00 | 03-36 | 0 | 90. | 0.467 | 9.30 | 4550 | 2.07 | 0.58 | 1024 | 3010 | 4.21 | 5.05 | 0 | 1.03 | . + | • |
| 0.00 0.004 0.0646 3.478 3180 3.47 1.338 792 2.264 4.002 10.879 0.0 0.005 0.275 0.00 0.0070 0.0086 0.518 3250 3.50 1.20 437 4.004 10.57 0.00 0.007 0.255 0.00 0.0070 0.0072 0.772 4.88 3250 3.50 1.25 930 2673 4.11 10.45 0.0 0.0 0.070 0.275 0.00 0.0070 0.0071 0.810 4.50 3.75 1.25 930 2673 4.11 10.45 0.0 0.0 0.710 0.275 0.00 0.0070 0.0074 1.360 4.50 3.175 1.25 930 2673 4.11 10.45 0.0 0.0 0.710 0.275 0.00 0.0070 0.0074 1.360 4.53 3.175 1.25 930 2673 4.11 10.45 0.0 0.0 0.710 0.275 0.00 0.0070 0.0074 1.360 4.53 3.175 1.25 930 2679 4.13 10.56 0.0 0.0 0.710 0.275 0.00 0.00 0.0074 1.360 4.53 3.175 1.26 4.37 2.391 4.02 2.0.61 0.0 0.710 0.0 0.775 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0. | 03-39 | • | • | 0.495 | 10.46 | 4950 | 2.14 | 0.53 | 1158 | 3240 | 4.24 | 4.62 | 0.0 | 66.0 | ~ | • |
| 1.00 | 04-03 | 0.0 | • | 0.645 | 3.49 | 2880 | 3,17 | 1.38 | 792 | 2264 | 4.02 | 10,89 | 0.0 | 69.0 | ^1 | • |
| 0.00 0.0092 0.772 6.18 3650 1.25 930 2637 4.11 10.45 0.0 0.0 0.70 0.70 0.70 0.70 0.70 0.70 | 07-10 |) · | ٠ | 0.686 | 4.18 | 3160 | 3.34 | 66.1 | 826 | 2662 | 603 | ္ပံ | 3 | 10.0 | | • |
| | 04-09 | 0 0 | • | 0.722 | 4.88 | 3250 | S (| 1.30 | 698 | 2437 | 4 . | ံ ေ | 0 0 | 0.0 | | • |
| 1.00 | 21-10 | • | • | 0.00 | 0• T 0 | 2040 | 0.00 | 1 25 | 876 | 6797 | 4.11 | , | • | 2 6 | | • |
| 1.00 | 04-15 | • | | 0.810 | 0.0 | 2765 | 000 | 1.25 | 200 | 6/07 | 4.13 | : - | • | 2.6 | | • |
| 1.00 | 11-03 | • • | 5 5 | 0.00 | 0647 | 4500 | 0.49 | 1.64 | † c | 2203 | 07.7 | • 6 | |) r | | |
| 1.09 | 11-05 | 000 | • | 1 420 | 400 | 3350 | 7.00 | 2.57 | 000 | 1467 | 4.04 | ; | | 100 | - | |
| 5-06 0.16 0.094 0.225 7.21 3400 1.11 0.41 835 2415 4.04 3.41 0.04 1.26 0.47 0.11 0.16 0.086 0.225 7.21 3400 1.11 0.41 858 2529 4.08 3.41 0.04 1.28 0.48 0.094 0.094 1.28 0.094 1.29 0.094 1.29 0.094 1.28 0.094 1.29 0.094 1.29 0.094 1.29 0.094 1.29 0.094 | 11-09 | 0 | • | 1.490 | 40.04 | 3650 | 7.18 | 2,43 | 948 | 2611 | 4,08 | | 0 | 0.43 | | |
| 509 0.16 6.088 0.225 7.21 3400 1.11 0.41 858 2529 4.08 3.41 0.04 1.43 0.57 0.51 512 0.016 0.028 0.229 7.21 3400 1.17 0.38 1058 2695 4.14 3.30 0.04 1.28 0.48 0.16 0.028 0.270 9.75 4.75 1.20 0.33 1098 2695 4.14 3.30 0.04 1.58 0.04 1.51 0.65 0.04 1.51 0.05 0.05 0.04 1.59 0.83 1098 2695 4.19 2.86 0.04 1.51 0.05 0.04 1.51 0.05 0.05 0.04 1.51 0.05 | 05-06 | , – | | 0.210 | 6.2.8 | 3175 | 1.06 | 0.41 | 83.5 | 2415 | 40.04 | 3.36 | 0.04 | 1.26 | + | 7 |
| 55-12 0.16 0.080 0.238 8.14 3750 1.17 0.38 898 2695 4.14 3.30 0.04 1.28 0.48 0.10 0.10 0.258 10.23 4575 1.20 0.33 1058 3018 4.18 2.86 0.04 1.37 0.51 0.10 0.16 0.005 0.258 10.23 4575 1.20 0.33 1100 3091 4.19 2.86 0.04 1.37 0.51 0.10 0.10 0.364 4.88 3000 1.30 0.78 812 2.31 2.81 0.04 1.37 0.55 0.10 0.10 0.364 4.88 3000 1.90 0.78 818 3.99 6.13 0.04 1.37 0.55 0.10 0.10 0.364 4.88 3000 1.90 0.78 818 3.99 6.13 0.04 1.37 0.35 0.10 0.10 0.364 4.88 3000 1.90 0.78 818 3.99 6.13 0.00 0.30 0.37 0.10 0.30 0.10 0.30 0.30 0.30 0.10 0.30 0.3 | 05-09 | _ | • | 0.225 | 7.21 | 3400 | 1.11 | 0.41 | 858 | 2529 | 4.08 | 3.41 | 0.04 | 1.43 | | 7 |
| 5-15 0.16 0.066 0.258 10.23 4575 1.20 0.33 1058 3018 4.18 2.86 0.04 1.37 0.51 0.51 0.16 0.063 0.270 9.76 4775 1.26 0.33 1100 3091 4.19 2.86 0.04 1.51 0.62 0.105 0.16 0.059 0.264 4.88 3000 1.90 0.78 858 2.18 3.99 6.13 0.04 1.37 0.05 0.25 0.10 0.364 4.88 3000 1.90 0.78 858 2415 4.02 6.06 0.08 1.00 0.37 0.10 0.106 0.009 0.286 4.88 3000 1.90 0.78 858 2415 4.02 6.06 0.08 1.00 0.37 0.10 0.10 0.10 0.10 0.10 0.10 0.10 0.1 | 05-12 | - | • | 0.238 | 8.14 | 3750 | 1.17 | 0.38 | 868 | 2695 | 4.14 | 3,30 | 0.04 | 1.28 | .+ | ٦. |
| 0.16 0.063 0.270 9.76 4775 1.26 0.33 1100 3091 4.19 2.86 0.04 1.51 0.62 0.15 5-21 0.16 0.059 0.283 12.55 5050 1.30 0.32 1143 3196 4.21 2.81 0.04 1.31 0.52 0.08 0.09 1.31 0.62 2.18 3.00 1.90 0.76 825 2.318 3.99 6.09 1.09 0.08 0.08 1.00 0.35 0.11 0.08 1.00 0.35 0.01 0.09 0.08 1.00 0.39 0.08 1.00 0.35 0.01 0.09 0.08 1.00 0.35 0.01 0.09 0.08 0.08 0.09 0.08 0.08 0.09 0.08 0.08 0.37 0.01 0.09 0.08 0.08 0.09 0.09 0.08 0.09 0.08 0.08 0.09 0.08 0.08 0.09 0.08 0.08 < | 05-15 | - | 90. | 0.258 | 10,23 | 4575 | 1.20 | 0.33 | 1058 | 3018 | 4.18 | 2.86 | 0.04 | 1.37 | in | ٦. |
| 0.16 0.059 0.283 12.55 5050 1.33 0.32 1143 3196 4.21 2.81 0.04 1.37 0.53 0.31 0.31 0.31 0.31 0.31 0.31 0.31 0.3 | 05-18 | _ | • | 0.270 | 9.16 | 4175 | 1.26 | 0.33 | 1100 | 3091 | 4.19 | 2.86 | 0.04 | 1.51 | 2 | • |
| 0.16 0.100 0.364 4.88 3000 1.90 0.78 825 2318 3.99 6.13 0.08 0.93 0.35 0.10 0.10 0.364 4.88 3000 1.90 0.78 825 2318 3.99 6.13 0.08 0.99 0.37 0.10 0.36 0.094 0.385 5.58 3200 1.97 0.76 858 2415 4.02 6.06 0.08 1.00 0.37 0.10 0.37 0.10 0.38 0.36 0.38 0.38 0.38 0.38 0.38 0.38 0.38 0.38 | 05-21 | ~ | • | 0.283 | 12.55 | 5050 | 1.30 | 0.32 | 1143 | 3196 | 4.21 | 2.81 | 0.04 | 1.37 | in | ٦. |
| 0.16 0.094 0.385 5.58 3200 1.97 0.76 858 2415 4.02 6.06 0.08 1.00 0.37 0.10 0.16 0.088 0.407 6.39 3400 2.35 0.82 917 2498 3.96 6.53 0.08 0.98 0.37 0.10 0.16 0.088 0.473 9.18 4400 2.09 0.62 1025 2950 4.19 5.40 0.07 1.01 0.39 0.10 0.16 0.063 0.473 9.18 4400 2.16 0.62 1025 2950 4.19 5.40 0.07 1.01 0.39 0.10 0.16 0.063 0.501 10.23 4750 2.28 0.60 1096 3079 4.20 5.23 0.07 1.01 0.39 0.10 0.16 0.0103 0.689 3.49 2700 3.24 1.18 749 2180 4.00 11.83 0.15 0.66 0.25 0.2 0.16 0.0103 0.680 0.766 6.04 3475 3.58 1.39 866 2269 4.00 11.02 0.14 0.74 0.27 0.2 0.16 0.080 0.808 6.97 3750 3.76 1.31 931 2673 4.13 10.61 0.13 0.77 0.28 0.2 0.16 0.092 1.605 0.850 7.90 4.25 3.87 1.21 966 2796 4.10 0.19 0.13 0.77 0.20 0.16 0.092 1.420 5.23 3450 6.95 2.48 922 2517 4.05 12.82 0.25 0.25 0.40 0.19 0.3 | 06-03 | ~ | • | 36 | 4.88 | 3000 | 1.90 | 0.78 | 825 | 2318 | 3,99 | 6.13 | 0.08 | 0.93 | ~ | 7 |
| 0.16 0.088 0.407 6.39 3400 2.35 0.82 917 2498 3.96 6,53 0.08 0.98 0.37 0.18 0.16 0.078 0.478 7.90 3850 2.09 0.68 925 2730 4.15 5.90 0.077 1.01 0.839 0.11 0.16 0.063 0.501 10.23 4750 2.16 0.62 10.25 29.04 4.20 5.23 0.07 1.01 0.39 0.11 0.16 0.011 0.663 0.501 10.23 4750 2.28 0.60 10.96 30.79 4.20 5.23 0.07 10.96 0.35 0.1 0.16 0.011 0.669 3.49 2.70 3.24 1.18 749 2180 4.00 11.83 0.15 0.25 0.25 0.25 0.25 0.25 0.25 0.25 0.25 0.25 0.25 0.25 0.25 0.25 0.25 0.25 | 90-90 | _ | • | 38 | 5.58 | 3200 | 1.97 | 0.76 | 858 | 2415 | 4.02 | 90.9 | 0.08 | 30. | ~ | ٦. |
| 0.16 0.078 0.448 7.90 3850 2.09 0.68 925 2730 4.15 5.90 0.07 1.05 0.39 0.10 0.10 0.078 0.248 7.90 3850 2.09 0.68 925 2730 4.15 5.90 0.07 1.05 0.39 0.10 0.16 0.008 0.473 9.18 4400 2.18 0.62 1025 2950 4.19 5.40 0.07 1.01 0.39 0.11 0.23 4750 2.28 0.65 0.62 1025 2950 4.20 5.24 0.07 1.01 0.39 0.11 0.23 4750 2.28 0.65 1.05 1.05 0.009 0.11 0.18 0.10 0.10 0.099 0.10 0.009 0.10 0.009 0.10 0.10 | 60-90 | _ | | 40 | 6.39 | 3400 | 2.35 | 0.82 | 917 | 2498 | 3.96 | 6,53 | 0.08 | 0.98 | ~ | ٦. |
| 0.16 0.068 0.473 9.18 4400 2.16 0.62 1025 2950 4.19 5.40 0.07 1.01 0.39 0.11 0.16 0.063 0.063 0.063 0.251 10.23 4750 2.28 0.60 1096 3079 4.20 5.23 0.07 0.96 0.35 0.15 0.16 0.011 0.689 3.479 2700 3.24 1.18 749 2180 4.00 11.63 0.15 0.46 0.25 0.2 0.25 0.2 0.26 0.25 0.2 0.16 0.103 0.869 4.07 2700 3.48 1.49 806 2296 4.00 11.63 0.15 0.46 0.25 0.2 0.25 0.25 0.2 0.16 0.094 0.710 4.88 3175 3.58 1.39 866 2396 4.02 11.02 0.14 0.74 0.27 0.25 0.36 0.25 0.36 0.25 0.36 0.25 0.36 0.25 0.36 0.25 0.36 0.25 0.36 0.25 0.36 0.25 0.36 0.25 0.36 0.25 0.36 0.25 0.36 0.25 0.36 0.25 0.38 0.25 0.38 0.26 0.25 0.38 0.35 0.36 0.387 1.21 966 2796 4.17 10.42 0.13 0.71 0.27 0.2 0.26 0.36 0.36 0.36 0.36 0.36 0.36 0.36 0.3 | 06-12 | ~ | ٠ | 44 | 7.90 | 3850 | 2.09 | 0.68 | 925 | 2730 | 4.15 | 2.90 | 0.07 | 1.05 | m | ٦. |
| 5-18 0.16 0.063 0.501 10.23 475C 2.28 0.60 1096 3079 4.20 5.23 0.07 0.96 0.36 0.10 0.10 0.10 0.10 0.23 475C 2.28 0.60 1096 3079 4.20 5.23 0.07 0.96 0.36 0.10 0.10 0.689 3.49 2700 3.24 1.18 749 2180 4.00 11.83 0.15 0.15 0.66 0.25 0.25 0.25 0.25 0.10 0.10 0.10 0.680 0.27 2900 3.48 1.19 806 2269 4.00 11.63 0.15 0.66 0.26 0.25 0.26 0.25 0.25 0.10 0.10 0.10 4.88 3175 3.58 1.39 806 2269 4.00 11.02 0.14 0.74 0.27 0.26 0.25 0.26 0.25 0.16 0.086 0.766 6.04 3475 3.78 1.34 909 2540 4.06 10.91 0.14 0.74 0.73 0.28 0.25 0.16 0.080 0.808 6.97 3750 3.76 1.31 931 2673 4.13 10.61 0.13 0.71 0.27 0.20 0.25 0.04 0.07 0.0850 7.90 4025 3.87 1.21 966 2796 4.17 10.42 0.13 0.71 0.20 0.28 0.20 0.16 0.092 1.420 5.23 3450 6.74 2.56 8.24 9.22 2517 4.03 20.15 0.26 0.25 0.44 0.19 0.3 | 06-15 | ~ | • | 47 | 9.18 | 4400 | 2.16 | 0.62 | 1025 | 2950 | 4.19 | 5.40 | 0.07 | 1.01 | m | ٦, |
| 5-21 0.16 0.111 0.639 3.49 2700 3.24 1.18 749 2180 4.00 11.83 0.15 0.15 0.66 0.25 0.25 0.25 0.16 0.103 0.680 4.07 2900 3.48 1.49 806 2269 4.00 11.63 0.15 0.15 0.68 0.26 0.25 0.25 0.25 0.10 0.10 0.094 0.10 0.10 0.10 0.10 0.10 0.10 0.10 0.1 | 06-18 | _ | ٠ | 50 | 10.23 | 475C | 2.28 | 0.60 | 1096 | 3079 | 4.20 | 5.23 | 0.07 | 96.0 | ~ | ٦, |
| 5-24 0.16 0.103 0.680 4.07 2900 3.48 1.49 806 2269 4.00 11.63 0.15 0.68 0.26 0.2 5-27 0.16 0.094 0.710 4.88 3175 3.58 1.39 866 2396 4.02 11.02 0.14 0.74 0.27 0.2 0.16 0.096 0.766 6.04 3475 3.78 1.34 909 2540 4.06 10.91 0.14 0.73 0.28 0.2 0.16 0.080 0.808 6.97 3750 3.76 1.31 931 2673 4.13 10.61 0.13 0.71 0.27 0.2 0.16 0.075 0.0450 7.90 4.025 3.48 1.21 966 2796 4.17 10.42 0.13 0.72 0.28 0.2 0.16 0.087 1.420 5.23 3450 6.95 2.48 922 2517 4.05 19.82 0.25 0.44 0.19 0.3 | 06-21 | _ | | 63 | 3.49 | 2700 | 3.24 | 1.18 | 642 | 2180 | 4.00 | 11,83 | 0.15 | 0.66 | O.I | .2 |
| 5-27 0.16 0.094 0.710 4.88 3175 3.58 1.39 866 2396 4.02 11.02 0.14 0.74 0.27 0.2 0.15 0.086 0.76 6.04 3475 3.78 1.34 909 2540 4.06 10.91 0.14 0.73 0.28 0.2 0.15 0.080 0.808 6.97 3750 3.76 1.31 931 2673 4.13 10.61 0.13 0.71 0.27 0.2 0.25 0.16 0.075 0.0850 7.90 4025 3.87 1.21 966 2796 4.17 10.42 0.13 0.77 0.2 0.28 0.2 0.16 0.092 1.360 4.65 3.250 6.74 6.55 2.848 9.22 2517 4.05 10.82 0.25 0.44 0.19 0.3 0.35 0.16 0.087 1.420 5.23 3450 6.95 2.48 9.22 2517 4.05 10.82 0.25 0.25 0.44 0.19 0.3 | 06-24 | $\overline{}$ | • | 68 | 4.07 | 2900 | 3.48 | 1.49 | 806 | 5269 | 4.00 | 11.63 | 0.15 | 0.68 | 01 | 2 |
| 5-30 0.16 0.086 0.766 6.04 3475 3.78 1.34 909 2540 4.06 10.91 0.14 0.73 0.28 0.2 5-33 0.16 0.080 0.808 6.97 3750 3.76 1.31 931 2673 4.13 10.61 0.13 0.71 0.27 0.2 5-36 0.16 0.075 0.850 7.90 4.025 3.87 1.21 966 2796 4.17 10.42 0.13 0.72 0.28 0.2 6.16 0.092 1.420 4.65 3.25 0.48 9.2 2517 4.05 10.82 0.25 0.44 0.19 0.3 | 06-27 | _ | | 7.1 | 4.88 | 3175 | 3.58 | 1.39 | 866 | 2396 | 4.02 | 11.02 | 0.14 | 0.74 | \sim | |
| 5-33 0.16 0.080 0.808 6.97 3750 3.76 1.31 931 2673 4.13 10.61 0.13 0.71 0.27 0.2 5-36 0.16 0.075 0.850 7.90 4.25 3.87 1.21 966 2796 4.17 10.42 0.13 0.72 0.28 0.2 6.16 0.092 1.360 4.65 3250 6.77 2.56 866 2425 4.03 20.15 0.26 0.51 0.19 0.3 6.16 0.087 1.420 5.23 3450 6.95 2.48 922 2517 4.05 10.82 0.25 0.44 0.19 0.3 | 06-30 | _ | • | 76 | 6.04 | 3475 | 3.78 | 1.34 | 606 | 2540 | 4.06 | 10.91 | 0.14 | 0.73 | \sim | 7 |
| 5-36 0.16 0.075 0.850 7.90 4025 3.87 1.21 966 2796 4.17 10.42 0.13 0.72 0.28 0.2 5-03 0.16 0.092 1.360 4.65 3250 6.74 2.56 866 2425 4.03 20.15 0.26 5.51 0.19 0.3 5-06 0.16 0.087 1.420 5.23 3450 6.95 2.48 922 2517 4.05 19.82 0.25 0.44 0.19 0.3 | .06-33 | - | • | 80 | 6.97 | 3750 | 3.76 | 1,31 | 931 | 2673 | 4.13 | 10.61 | 0.13 | 0.71 | \sim | ٠, |
|)-03 | 06-36 | - | .07 | 95 | 7.90 | 4025 | 3.87 | 1.21 | 996 | 2796 | 4.17 | 10.42 | - | 0.72 | O 1 | . 2 |
|)-06 | 10-03 | ~ | • | 36 | 4.65 | 3250 | 6.74 | 5.56 | 998 | 2425 | 4.03 | 20.15 | 2 | 0.51 | _ | ٠. |
| | 10-06 | _ | | • | | | | | | 1 | | | 1 | | | |

^{*}Free-stream conditions based on frozen flow model.
**Boundary-layer predictions based on nonequilibrium flow-equilibrium catalytic wall conditions.

TABLE I.- TEST CONDITIONS FOR HEAT-TRANSFER MEASUREMENTS ON MODEL I - Concluded

| P ₀ , H ₀ , atm MJ/kg |
|--|
| 183 |
| 202 |
| 214 |
| 0.224 8. 0.246 9. |
| 259 11 |
| 273 12. |
| 365 4. |
| 380 5 |
| 595 0. 631 7. |
| 6 25 |
| 484 9. |
| 640 3. |
| 680 4. |
| 111 |
| 797 6. |
| 832 7 |
| 360 |
| 500 |
| 178 3. |
| 207. 5. |
| 226 6. |
| 254 |
| 265 11. |
| 276 13. |
| 358 4. |
| 380 5. |
| 204 |
| 8 0/4 |
| 500 10. |
| 640 3 |
| 9 0 0 1 |
| 017 |
| 097 |
| 000 |
| |
| |

**Boundary-layer predictions based on nonequilibrium flow-equilibrium catalytic wall conditions. *Free-stream conditions based on frozen flow model.

Table II.- Test conditions for pressure measurements on model i

| O N O | E L T I O N S | S T A | G N A T D I T I | N 0 I 0 N S | ⊬∢ | R E E S H H | F S T | A E P C | L 0 C / | 1 T I O N I O N | v> ∗ | O L L | E S | LAYER TEP** |
|------------------------------|------------------|-------|---------------------------|------------------|--|---|----------|--------------|------------|--|---------------------------------------|----------|------|---|
| $T_{\rm w}/T_0$ $^{\rm p_0}$ | atm | _ | H ₀ , MJ/kg | T ₀ , | $\begin{array}{c} ^{\mathrm{p}_{\infty'}} \\ \mathrm{atm} \\ \times 103 \end{array}$ | $\begin{array}{c} \rho_{\infty}, \\ \text{kg/m}^3 \\ \times 10^3 \end{array}$ | F, & X | u∞, m/sec | × × | $^{ m N_{Re}_{\infty,L}}_{	imes10^{-3}}$ | $^{ m NRe}_{\infty,h} \times 10^{-3}$ | δ, cm | cm, | у/ч |
| 760 | C.37 | ۵ | 5,58 | 3200 | 1.95 | 0.75 | 860 | 2415 | 4.02 | 5.98 | 0.0 | 1.02 | 0.40 | 0.0 |
| 376 | 0.439 | _ | 8.14 | 3925 | 2. 04 | 0.65 | 954 | 2762 | 4.15 | 5.68 | 0.0 | 96.0 | 0.37 | 0.0 |
| 860 | 0.697 | | 4.53 | 3075 | 3,53 | 1.42 | 845 | 2349 | 00°4 | 11.15 | 0.0 | 0.72 | 0.28 | 0.0 |
| 086 | 0. 780 | | 9.10 | 3500 | 2.5 | 1.34 | 016 | 2553 | 10.4 | 10.97 | ٥ • | 5 6 | 7.0 | |
| 480 | 1.590 | | 2 6 6 | 3775 | 61.7 | 2.47 | 956 | 2669 | 4.10 | 20.30 | 000 | 0.51 | 0.19 | |
| 081 | 1.905 | | 5.81 | 3700 | 9.39 | 3.11 | 971 | 2626 | 4.06 | 25.07 | 0.0 | 0.49 | 0.18 | 0.0 |
| 093 | 0.380 | | 5.81 | 3225 | 1.95 | 0.74 | 862 | 2428 | 4.02 | 5.96 | 60 • 0 | 1001 | 0.39 | 0.195 |
| 375 | 0.432 | | 8.25 | 4300 | 2.01. | 0.63 | 646 | 2793 | 4.16 | 5.50 | 60 0 | 1.06 | 0.42 | 0.187 |
| 0.98 | 0.700 | | 4.42 | 3050 | 3,52 | 1.43 | 838 | 2339 | 4.01 | 11.05 | 0.18 | 0.72 | 0.27 | 0.274 |
| 086 | 0.785 | | 40.0 | 3475 | 3,90 | 1,38 | 216 | 2538 | 4.05 | 11.21 | 0°18 | 2.0 | 7.70 | 0.275 |
| 0.081 1.640 | 1.640 | | 6.04 | 3775 | 7.78 | 2,59 | 959 | 2645 | 00.4 | 21.19 | 0.34 | 0.51 | 0.19 | 0.392 |
| 080 | 0.380 | | 6.04 | 3750 | 1.83 | 0.61 | 913 | 2685 | 4.12 | 5.22 | 0.21 | 1.09 | 0.43 | 0.467 |
| 070 | 0.428 | | 4.07 | 4275 | 2.00 | 0.59 | 1004 | 2901 | 4.16 | 5.10 | 0.21 | 1.10 | 0.43 | 0.463 |
| 102 | 0. 100 | | 4.18 | 2950 | 3.58 | 1.50 | 818 | 2291 | 4.00 | 11.74 | 0.48 | 0.71 | 0.27 | 0.715 |
| 087 | 0.788 | | 5.93 | 3450 | 3,86 | 1,39 | 905 | 2528 | 4.06 | 11,23 | 0.46 | 69.0 | 0.26 | 0.734 |
| 980 | 1.520 | | 2.35 | 3500 | 87.7 | 2.58 | 126 | 2545 | 4.06 | 20.02 | * a | 2 0 | | 1.029 |
| 087 | 1.960 | | 2.00 | 3450 | 10.27 | 3,60 | 948 | 2504 | 3,99 | 28, 10 | 1,14 | 0.40 | 0.15 | 1.104 |
| 395 | 0.389 | | 5.99 | 3250 | 2.01 | 0.76 | 869 | 2439 | 4.02 | 60.9 | 0.35 | 1.00 | 0.39 | 0.711 |
| 079 | 0.447 | | 7.90 | 3800 | 2.11 | 0.70 | 921 | 2688 | 4.14 | 5.98 | 0.34 | 1.05 | 0.42 | 0.680 |
| 160 | 0.700 | | 4.65 | 3100 | 3.57 | 1.42 | 852 | 2360 | 4.01 | 11.15 | 0.63 | 0.72 | 0.27 | 0.985 |
| 0.86 | 0.788 | | 6.04 | 3475 | 3,91 | 1.39 | 915 | 2538 | 4.05 | 11.25 | 0° 64 | 0.72 | 0.27 | 0.993 |
| 787 | 1.510 | | 5.43 | 2420 | 1.04 | 60.7 | 776 | 1167 | 000 | 20.30 | 1.19 | 0.0 | 5.0 | 1000 |
| 200 | 1.075 | | 87.0 | 3130 | | 2007 | 600 | 2007 | 900 | 2000 | 0 ° 7 ° 1 | 70.0 | | 1 535 |
| 781 203 | 1.925 | | 1 0 0 U | 27.6 | 4.65 | 36 | 200 | 2040 | 800 | 20.0 | 7 * * 0 | • | • | 1,000 |
| 27.5 | 0.00 | | | 75.70 | 1.00 | | 100 | 11.00 | 0 . | | 0 0 | 1 6 | | |
| 9 0 | 0 1 1 0 | |) c | 4167 | 70.7 | 70.0 | 700 | 7047 | 100 | 7.07 | • | 100 | | P 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 |
| 200 | 0,000 | | 7 * * * | 2020 | 0000 | 74. | ο c | 6557 | | 11.11 | 1,0 | 2 5 | 7.0 | 1.44 |
| 086 | 087.0 | _ | 0.04 0.04 | 3475 | 7.00 | 1.38 | 716 | 25.38 | 4. U.O. | 11.15 | 7. 0. | 5.0 | 27.0 | 1.040 |
| 084 | 1.500 | | 2 . | 35(5 | 61. | 2° 43 | 938 | 1167 | 20.0 | 20.13 | 1.04 | 20.0 | | 7.000 |
| 080 | 1.090 | | 0 0 | 3130 | 900 | 2.03 | 707 | 2602 | φ. | **** | 1.00 1.03 | 70.0 | | 7 003 |
| \circ | 1.900 | | 2.28 | 3650 | ٧• ١٥ | 3.10 | 7.0 8 | 9097 | *** | 24.90 | Z• U3 | * | 21.0 | <.vo |

**Boundary-layer predictions based on nonequilibrium flow-equilibrium catalytic wall conditions. *Free-stream conditions based on frozen flow model.

TABLE III.- TEST CONDITIONS FOR HEAT-TRANSFER MEASUREMENTS ON MODEL II

| | : | | | | - | י ב | - T | | | | - c | | | - | , | |
|---------------|-------------|-------------|-----------------|--------------------------|---|--------|---------------------|---|--------|--------------|------------|--------------------------------|-------------------------------|----------|---------------------------------------|--------|
| | h, cm | ≱ | $T_{\rm w}/T_0$ | P ₀ , atn, | H ₀ , MJ/kg | T. W. | pمر atm × 103 | $ ho_{\infty}^{ ho_{\infty}}$ kg/m ³ $	imes$ 10 ³ | F, 87A | n∞, m/sec | ≥ 8 | ^N Re ω, L × 10-3 | N _{Re ω,h} × 10-3 | δ, cm | cm, | р/б |
| 0-01 | 0.0 | 0.0 | 0.082 | 22 | 1 6 | 3650 | 1.13 | 60 | 894 | 645 | 4.10 | 2.61 | 0.0 | | 0.45 | 0.0 |
| 10-0 | ٥•٥ | • | 0.063 | 25 | ď | 4750 | 1.24 | 3 | 1110 | 910 | 4.15 | 2.19 | 0.0 | | 0.45 | 0.0 |
| 10-1 | 0 | • | 0.092 | 38 | æ | 3250 | 5.06 | ~ | 877 | 435 | 3.99 | 4.87 | 0.0 | | 0.30 | 0 |
| 1-0-1 | ٥ • • | • | C.C78 | ₹. | σ, | 3850 | 2.24 | ► . | 943 | 722 | 4.10 | 4.87 | 0.0 | | 0.31 | 0 0 |
| 10-1 10-1 | ٠ • • | • | 0.069 | 47 | ~ (| 4375 | 2.32 | ب و | 1038 | 933 | 4.18 | 4.47 | 0.0 | | 0.33 | 0 0 |
| 2-0-1 10-2 | • | ء د د | 0.103 | 9 6 | ^ 0 | 27.50 | 200 | n 4 | 679 | 797 | 0,00 | 3, 6 | • | | 0,22 | |
| 10-2 | | | 180-0 | - 6 | • • | 3725 | 4.05 | 7 | 776 | 55.5 | 4.08 | 8.78 | | | 0.23 | 0 |
| 10-3 | 0.0 | 0.0 | 0.077 | 87 | • 🛨 | 3900 | 4.24 | ~ | 965 | 735 | 4.11 | 8.99 | 0.0 | | 0.23 | 0.0 |
| 0-50 | 09 0 | C. 119 | 0.081 | 22 | 9 | 3694 | 1.13 | 3 | 905 | 663 | 4.09 | 2.55 | 0.15 | | 0.49 | 0.487 |
| 0-60 | 0.00 | | 0.082 | 22 | Ð. | 3650 | 1.11 | 3 | 894 | 645 | 4.10 | 2.57 | 0.15 | | 0.42 | 0.534 |
| 0-40 | 0.60 | C. 708 | 0.063 | 24 | 3 | 4800 | 1.16 | 3 | 1116 | 960 | 4.16 | 2.04 | 0.12 | | 0.44 | 0.493 |
| 0-60 0-60 | 0.60 | | 0.063 | 25 | - 1 | 4775 | 1.24 | ന ദ | 1116 | 0.84 | 4.15 | 2.17 | 0.13 | | 0.46 | 0.491 |
| 04-1 | 0.00 | 0.613 | 2600 | 3 2 | S C | 3166 | 1.91 | ~ * | 862 | 395 | 3,99 | 4.09 | 0.28 | | 0 0 | 0.721 |
| 90-09-12 | 0.00 | | 6000 | 0.48 | 7 ° ° ° ° ° ° ° ° ° ° ° ° ° ° ° ° ° ° ° | 3225 | 2,10 | 0.40 | 8 Z 8 | 2427 | 3, 98 | . 6 | 0.30 | 0.87 | 9 6 | 0.687 |
| 04-1 | 09.60 | S | 0.084 | 77 | 2 | 3560 | 2.10 | ٠ ٢- | 206 | 589 | 4.06 | 4.76 | 0.29 | | 0.30 | 0.728 |
| 05-1 | 0.60 | 0.593 | 0.081 | 3 | Š | 3700 | 2.16 | ~ | 921 | 959 | 4.08 | 4.82 | 0.29 | | 0.31 | 0.712 |
| 09-1 | 09.0 | | 0.078 | 43 | 0 | 3850 | 2.19 | ~ | 643 | 723 | 4.10 | 4.73 | 0.29 | | 0,31 | 0.720 |
| 05-1 | 0.60 | 0.512 | 0.070 | ÷. | 0 | 4315 | 2.23 | so v | 1025 | 911 | 4.14 | 4.37 | 0.26 | | 0.0 | 0.694 |
| 2412 | 0000 | 2 6 | C. 102 | 4 0 | > c | 2928 | 30.06 | + u | 178 | 010 | 0.4°C | 8.40 | 0.0 | | 0.22 | 1.004 |
| 2-60 | 09.60 | ٦. | 0.103 | 9 9 | ס כ | 2900 | 3.76 | ľ | 819 | 0 2 2 | 3.96 | 74.6 | | | 0.22 | 1.031 |
| 04-2 | 09.0 | ~ | 0.087 | 72 | · on | 3437 | 3.80 | 'n | 919 | 515 | 4.01 | 8.50 | 0.51 | | 0.23 | 0.969 |
| 04-2 | 09.0 | 0.719 | 0.082 | 16 | ~ | 3657 | 3.89 | 3 | 941 | 622 | 4.05 | 8.44 | 0.51 | | 0.23 | 0.960 |
| 09-2 | 0.60 | | 0.087 | 7 | æ | 3450 | 4.06 | 3 | 922 | 520 | 4.01 | 9.03 | 0.54 | | 0.22 | 666 0 |
| 05-2 | 0.60 | 0.786 | 0.086 | 7 | 0 | 3470 | 4.03 | , 。 | 923 | 530 | 4.01 | 8.96 | 0.54 | | 0.22 | 566.0 |
| 2-60 | 0, 60 | 0.00 | 0.083 | | ო. | 3600 | 60.4 | . * * | 930 | 596 | 90. | 9.05 | 0.55 7.1 | | 0.22 | 0.00 |
| 2 4 - 3 | 00.0 | | 0.077 | 9 5 | . | 3875 | 3.96 | v | 826 | 527 | 11: | \$. \$. | 15.0 | | 0.23 | 0 |
| 2000 | 0.00 | • | | ລຸດ | \$ C | 3006 | 4.22 | 7 | 200 | (3) | 11.00 | 9. 4. c | 0.04 | | 0.63 | 10,404 |
| 2 6 | 1.02 | | 2000 | ,, | | 26.75 | 1.13 | ٦ ~ | 0 0 | 7 7 7 | · · | 2,53 | 25.0 | | 0 0 | 0.820 |
| 7-10 | 1.02 | 0.224 | 200.0 | 7 7 | rα | C105 | 0 0 | ∩ r | 872 | 100 | 00 | 6.49 | 0.46 | | | 1.130 |
| 08-1 | 1.02 | 6 | 460°0 | 3 | ຸເຄ | 3185 | 2.04 | . ~ | 867 | 403 | 3.98 | 4.88 | 0.50 | | 0.34 | 1.153 |
| 07-1 | 1.02 | 0.284 | 0.084 | 6 | _ | 3580 | 2.09 | • | 912 | 598 | 4.05 | 4.68 | 0.48 | | 0.31 | 1.212 |
| 03-0 | 1.02 | • | 9.000 | 45 | _ | 3958 | 2.11 | · | 096 | 169 | 4.11 | 4.47 | 94.0 | | 0.31 | 1,134 |
| 0-07-1 | 1.02 | 0.264 | 0.068 | 43 | m | 4405 | 2.08 | Ś | 1042 | 946 | 4.14 | 3.99 | 0.41 | | 0.34 | 1,115 |
| 0-08-1 | 1.02 | 0.0 | 0.078 | €. | ο. | 3846 | 2.17 | ┡. | 945 | 721 | 4.10 | 4.70 | 0.48 | | 0.31 | 1.225 |
| 0-08-1 | 1.02 | 0 8 | 0.069 | 9 (| ο. | 4320 | 2.26 | • | 1026 | 913 | 51. | 4.39 | 0.45 | | 0.32 | 1.186 |
| 1-00 | 70. | 0.337 | 550 | 3 | s t | 3030 | 4.45 C | η, | 9 6 | 363 | 3.40 | 77.0 | 000 | | 0.60 | 7007 |
| 7-10-0 | 70.1 | * 7 | 0.100 | ö | ₹ | 0667 | 7.47 | \$ | 936 | 302 | | 000 | 60.0 | | \ \ \ \ \ \ \ \ \ \ \ \ \ \ \ \ \ \ \ | 1007 |
| | - | c | 30. | ٩ | • | ,000 | 36.0 | u | | | , | 17 | | | | 1 76 |

*Free-stream conditions based on frozen flow model.

**Boundary-layer predictions based on nonequilibrium flow-equilibrium catalytic wall conditions.

TABLE III.- TEST CONDITIONS FOR HEAT-TRANSFER MEASUREMENTS ON MODEL II - Concluded

| N N | 000 | MODEL CONDITION | S N O | 0 N O N O | 6 N A T I D I T I O | N S | 7 | E E S T H E | X N N A H | E P C () | 0 Z | 0 * 2 Z 2 O 1 I | | 8 A ⊃ - × × × × × × × × × × × × × × × × × × | H E X | T E P + E |
|-----------|----------|--------------------|-------|-------------------------|---------------------------|------------------|--|---|-----------------|--------------|--------|--|---|--|------------|-----------|
| | cm cm | 3 | Tw/T0 | P ₀ , atm | н ₀ , МЈ/кg | T ₀ , | $\begin{array}{c} p_{\infty},\\ \text{atm}\\ \times 10^3\end{array}$ | $k_{\rm g/m}^{ ho_{\infty'}} \times 10^3$ | H & | u∞, m/sec | M 8 | $^{ m N}_{ m Re}{}_{\infty, m L}$ $	imes 10^{-3}$ | $^{ m N}_{ m Re}{}_{\infty, m h} 	imes 10-3$ | δ, cm | δ*, cm, | у/ч |
| 0-07-24 | 1.02 | 0.358 | 0.088 | 0.714 | 5.88 | 3420 | 3.74 | 1,33 | 915 | 2508 | 4.01 | 8,42 | 0.86 | 0.62 | 0.23 | 1.642 |
| 0-02-18 | 1.02 | 0.407 | 0.089 | 0.750 | 5,58 | 3375 | 3.95 | 1.42 | 911 | 2485 | 4.00 | 8.95 | 0.92 | 0.60 | 0.22 | 1.701 |
| 190-01-27 | 1.02 | 0.396 | 08000 | 0.755 | 6.97 | 3732 | 3.87 | 1.27 | 156 | 2656 | 4.06 | 8.28 | 0.85 | 0.63 | 0.23 | 1.663 |
| 0-08-24 | 1.02 | 0.0 | 0.087 | 0.766 | 5.81 | 3440 | 4.02 | 1.42 | 920 | 2516 | 4.01 | 8.98 | 0.92 | 09.0 | 0.22 | 1.701 |
| 0-07-30 | 1.02 | 0.435 | 0.076 | 0.802 | 7.67 | 3936 | 3.91 | 1.23 | 896 | 2751 | 4.11 | 8.23 | 0.84 | 0.63 | 0.24 | 1.610 |
| 0-03-12 | 1.02 | 0.491 | 0.079 | 0.805 | 7.15 | 3775 | 3.88 | 1.28 | 943 | 2681 | 4.10 | 8.48 | 0.87 | C. 62 | 0.23 | 1.634 |
| 0-08-27 | 1.02 | o. 0 | 0.080 | 0.815 | 6.97 | 3750 | 4.02 | 1,32 | 946 | 2667 | 4.08 | 8.71 | 0.89 | 0.61 | 0.23 | 1.675 |
| 8-05-09 | 1.02 | 0.442 | 0.077 | 0.865 | 7.56 | 3900 | 4.20 | 1,34 | 965 | 2735 | 4.11 | 8.90 | 0.91 | 0.61 | 0.23 | 1.667 |
| 0-08-30 | 1.02 | 0.0 | 0.077 | 0.870 | 7.56 | 3920 | 4.25 | 1.34 | 016 | 2742 | 4.10 | 8.94 | 0.92 | 0.61 | 0.23 | 1.658 |

*Free-stream conditions based on frozen flow model.

**Boundary-layer predictions based on nonequilibrium flow-equilibrium catalytic wall conditions.

TABLE IV.- TEST CONDITIONS FOR PRESSURE MEASUREMENTS ON MODEL II

| $ \begin{array}{cccccccccccccccccccccccccccccccccccc$ | R U N | 0 0 | MODELNDITI | S N O | STAG | N A T I | N S | FR | E E S T F T H E | R E A M S → E | C 0 | N D I T O C A T | S N O I | 8 A | 0 U N D | A R Y | LAYE TEP* |
|--|-----------|----------|------------|----------------|-------------------------|---------------------------|------------------|--|---|------------------|-------------|--|---------|---|---------|-------------------|--------------|
| 97-05-06 0.0 0.0 0.001 0.0236 9.24 4250 1.118 0.34 1009 2889 4.112 2.33 0.0 0.119 0.05-06 0.0 0.0 0.002 0.00 | | h, cm | ≱ | 1 | P ₀ , atm | Н ₀ , МJ/kg | T ₀ , | $\begin{array}{c} ^{p_{\infty}},\\ \text{atm}\\ \times10^3\end{array}$ | $\begin{array}{c} \rho_{\infty}, \\ \text{kg/m}^3 \\ \times 10^3 \end{array}$ | F,8× | u, m/sec | 8 | ήü | $^{ m N}_{ m Re}{}_{\infty, m h} \times 10^{-3}$ | | co.* | h/8 |
| 97-95-12 | 197-05-06 | 0.0 | 0.0 | 0.071 | 23 | 9.24 | 4250 | 1.8 | 1 ~ | • | | 4.12 | 2.33 | 0.0 | 1.19 | 1 4 | 0.0 |
| 9-05-15 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0. | 197-05-12 | 0.0 | ٥ ٥ | 0.094 | 36 | 5.52 | 3175 | 46. | ~ . | ١٥. | | 3.98 | 4.65 | 0.0 | 0.82 | 0.30 | 0.0 |
| 97-65-21 0.0 0.0 0.101 0.0708 4.18 2940 1.52 813 2.520 1.50 0.0 0.00 0.00 0.00 0.00 0.00 0.00 | 197-05-15 | 000 | | 0.070 | 4 5 5 | 9.14 | 4300 | 87. | ^ | | | 4.11 | 4.45 | 000 | 0.86 | ~ ~ | 000 |
| 97-05-24 0.0 0.0 0.0 0.0087 0.170 5.93 3450 4.06 1.43 972 222 0.00 4.01 0.00 0.00 0.00 0.0087 0.1423 4.18 1375 97.2 22.2 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0 | 97-05 | 0.0 | 0 | 0.101 | 68 | 4.18 | 2960 | 6.68 | 10 | | | 3,96 | 9.28 | 0.0 | 0.58 | 2 | 0.0 |
| 97-05-30 0.0 0.00 0.007 11423 4.18 3080 4.16 1137 0.29 31 3.96 18.38 0.00 0.42 197-05-30 0.0 0.00 0.0082 11570 5.81 3.967 3.408 1157 3.09 0.81 2.61 8.38 0.00 0.42 197-05-30 0.0 0.0 0.00 0.0082 11570 5.81 3.967 3.408 2.65 3.40 18.38 0.00 0.44 196.89 0.00 0.00 0.00 0.00 0.00 0.00 0.00 0 | 97-05 | 0 | 0 | 0.087 | 77 | 5.93 | 3450 | 90. | + 1 | ~ . | | 4.01 | 9.03 | 0.0 | 0.60 | 2 | 0.0 |
| 91-05-36 | 97-05 | 0 0 | 0 0 | 0.081 | B. | 6.80 | 3700 | • • | ~ ~ | | | 4.06 | 8,91 | 0.0 | 0.61 | ~ ~ | 000 |
| 97-01-05 | 97-05 | 0 | | 0.082 | 5,4 | 5.83 | 3675 | 0 | · • | - ac | | 4.04 | 16.30 | | 0.44 | - | |
| 97-01-02 10.02 (2.276 0.082 0.226 7.90 3550 1.13 0.39 694 25.99 4.10 2.10 3.11 0.32 1.10 0.39 1.10 0.20 1.10 0.20 0.20 0.20 0.20 0.20 | 97-05 | 0.0 | 0.0 | 0.085 | 89 | 5.35 | 3550 | 69 | ~ | m | | 4.02 | 20.78 | 0.0 | 0.46 | | 0.0 |
| 997-01-13 1.02 C.237 0.0594 0.240 6.52 3200 1.28 0.49 656 2419 4.00 3.11 0.32 1.10 0.32 1.00 0.31 0.240 0.5242 0.324 0.3240 1.29 0.349 0.379 0.3 | 97-04 | 1.02 | 0.276 | 0.082 | 22 | 7.90 | 3650 | .13 | ~ | | | 4.10 | 2.61 | 0.27 | 1.23 | 4 | 82 |
| 97-701-12 1.00 0.01 0.362 8.37 2.990 0.75 8.45 2.70 0.77 0.31 9.70 0.362 0.79 0.79 8.45 2.70 0.70 | 97-01 | 1.02 | C.237 | 0.094 | 54 | 6.22 | 3200 | •28 | + 1 | Δ. | | 00.4 | 3,11 | 0.32 | 1.10 | 4. | 0.926 |
| 97-03-12 | 20-76 | 1.02 | 0.0 | 20.00 | 77 | 8.37 | 3900 | 200 | ~ 0 | Λ. | | 4.12 | 2.62 | 0.27 | 1.10 | 4 (| 92 |
| 97-04-12 1.02 0.384 0.094 0.368 5.70 3200 2.00 0.76 872 2410 3.98 4.75 0.49 0.80 0.81 0.99 0.905 0.422 6.97 350 2.12 0.74 900 2.827 4.07 4.87 0.69 0.81 0.81 0.905 0.422 6.97 350 2.15 0.74 900 2.827 4.07 4.87 0.69 0.81 0.81 0.905 0.422 0.431 7.90 3825 2.15 0.69 937 2713 4.10 4.60 0.48 0.88 0.99 0.905 0.430 | 97-03 | 1.02 | | 101.0 | 3 6 | - C | 3050 | 300 | nr | + . | | 5.45 4.06 | 70.4 | 0.50 | 0 0 | ソィ | 2,6 |
| 97-01-15 1.02 6.386 6.085 0.422 6.97 3530 2.12 0.74 90 2577 4.07 4.81 0.59 0.83 97-04-15 1.02 6.040 0.082 0.425 7.32 3850 2.14 0.673 9.67 4.81 0.499 0.849 97-04-15 1.02 0.040 0.475 0.453 9.13 4350 2.14 0.62 1033 2924 3.98 4.25 0.84 0.84 97-01-18 1.02 0.040 0.475 0.453 9.71 4.020 2.21 0.64 101 2.64 0.64 0.44 0.67 0.44 0.67 0.44 0.67 0.44 0.67 0.44 0.67 0.44 0.67 0.44 0.67 0.44 0.67 0.44 0.67 0.44 0.67 0.44 0.67 0.44 0.67 0.44 0.67 0.44 0.67 0.44 0.67 0.64 0.67 0.64 0 | 97-04 | 1.02 | 0.384 | 0.094 | 36 | 5.70 | 3200 | 00 | - 1 | ۱ ۸۰ | | 3.98 | 4.75 | 0.49 | 08.0 | ח ע | 2 2 |
| 97-03-15 1.02 | 97-01 | 1.02 | 0.386 | 0.085 | 42 | 26.91 | 3530 | •15 | _ | _ | | 4.07 | 4.87 | 0.50 | 0.81 | 3 | 24 |
| 97-04-15 1.02 | 97-03 | 1.02 | 0.0 | 0.082 | 45 | 7.32 | 3650 | •15 | - | ν. | | 4.07 | 4.81 | 64.0 | 0.83 | 3 | 22 |
| 97-04-18 1.02 0.474 0.075 0.455 9.07 4300 2.21 0.64 1021 2906 4.14 4.34 0.44 0.88 97-04-18 1.02 0.474 0.075 0.455 9.07 4300 2.22 0.64 1021 2906 4.14 4.34 0.44 0.88 97-04-18 1.02 0.474 0.070 0.455 9.07 4300 2.22 0.64 1021 2906 4.14 4.34 0.44 0.88 197-04-18 1.02 0.474 0.070 0.455 9.07 4300 2.22 0.64 1021 2906 4.14 4.34 0.44 0.88 197-04-21 1.02 0.383 0.101 0.680 3.84 2825 3.68 1.56 82 227 3.96 9.39 0.96 0.88 0.89 97-04-21 1.02 0.383 0.101 0.680 4.18 2960 3.68 1.56 82 227 3.96 9.28 0.96 0.55 99-07-04-21 1.02 0.383 0.101 0.680 4.18 2960 3.68 1.55 83 2290 3.96 9.28 0.96 0.55 99-07-04-21 1.02 0.381 0.092 0.765 5.46 3.350 4.05 1.42 9.20 2.22 9.20 9.28 0.99 9.28 0.99 9.29 0.99 9.29 0.99 9.29 0.29 9.29 0.29 9.29 0.29 9.29 0.29 9.29 0.29 9.29 0.29 9.29 0.29 9.29 0.29 9.29 0.29 9.29 0.29 9.29 0.29 9.29 0.29 9.29 0.29 9.29 0.29 9.29 9 | 97-04 | 1.02 | 6.430 | 6,078 | 43 | 7.90 | 3825 | 4. | Δ, | | | 4.10 | 4.66 | 0.48 | 0.84 | m (| 2: |
| 97-04-18 1.02 0.474 0.070 0.455 9.07 4300 2.21 0.64 1021 2975 4.12 4.02 0.44 0.45 9.79 1.02 0.476 0.47 | 7-01 | 1.02 | 194.0 | 0.00 | φ. γ | 9.13 | 4350 | | n u | n c | | 3,98 | 4.25 | 44.0 | 88.0 | ro r | 14 |
| 97-01-21 1.02 0.0 0.103 0.675 4.07 2920 3.68 1.54 825 2271 3.95 9.39 0.96 0.59 977-01-21 1.02 0.385 0.106 0.680 3.84 2825 3.68 1.60 798 2229 3.96 9.78 1.00 0.589 977-01-21 1.02 0.381 0.021 0.482 0.482 4.02 1.52 893 2424 3.98 9.42 0.995 0.589 977-01-24 1.02 0.383 0.000 0.765 5.46 3350 3.91 1.43 900 2472 4.02 9.02 0.993 0.599 977-01-24 1.02 0.000 0.768 5.93 3450 4.06 1.52 893 2426 3.98 9.40 0.995 0.599 977-01-24 1.02 0.001 0.0090 0.768 5.93 3450 4.06 1.42 922 2520 4.01 9.01 0.90 0.999 0.99 | 97-04 | 1.02 | 0.474 | 0.00 | 4 4 | 0.0 | 4300 | 21 | രം | | | 4.14 | 4.34 | 0 4 4 0 | 0.81 | 9 | 7 2 |
| 97-01-21 1.02 0.385 0.106 0.6680 3.84 2825 3.68 1.60 798 2229 3.96 9.78 1.00 0.568 97701-21 1.02 0.383. 0.101 0.680 4.18 2960 3.68 1.52 832 2290 3.96 9.28 0.95 0.589 0.701-24 1.02 0.391 0.0270 0.765 5.46 3350 3.91 1.43 900 2472 4.02 9.02 0.992 0.599 0.702-24 1.02 0.090 0.765 5.46 3350 3.91 1.43 900 2472 4.02 9.02 0.993 0.599 0.703-24 1.02 0.010 0.090 0.768 5.93 3450 4.05 1.44 922 2520 4.01 9.01 0.992 0.599 0.703-24 1.02 0.0417 0.086 0.800 5.999 3455 4.02 922 2520 4.02 9.28 0.995 0.599 0.703-24 1.02 0.010 0.084 0.810 6.34 3575 4.02 1.44 928 2832 4.02 9.28 0.995 0.599 0.703-24 1.02 0.0379 0.081 0.810 6.34 3575 4.02 1.44 928 2832 4.02 9.28 0.995 0.999 0.703-24 1.02 0.001 0.084 0.810 6.34 3575 4.02 1.44 924 2.81 4.06 8.88 0.91 0.61 0.899 0.703-24 1.02 0.001 0.090 1.495 4.88 3350 7.76 2.82 921 2464 4.00 1.744 1.79 0.44 0.704-30 1.02 0.402 0.402 0.087 1.505 5.23 3460 7.87 2.82 921 2464 4.00 1.744 1.79 0.44 0.704-34 1.02 0.481 0.088 1.585 5.23 3460 7.89 2.89 9.02 2.99 9.00 1.78 0.44 97-01-30 1.02 0.482 0.083 1.585 5.28 350 8.01 2.01 9.02 2.99 9.00 1.71 0.00 1.71 0.000 0.085 1.585 5.28 350 9.95 3.41 960 2599 4.00 1.71 0.00 1.71 0.0045 0.703-34 1.02 0.008 1.595 5.75 3630 9.95 3.41 960 2599 4.00 2.100 0.0045 0.703-34 1.02 0.008 1.595 5.11 3480 9.90 3.45 951 2518 4.00 2.100 0.0045 0.0085 1.900 0.82 3.41 960 2599 4.00 2.100 0.0045 0.0086 1.905 5.11 3480 9.90 3.45 951 2518 4.00 2.100 0.0045 0.0045 0.0086 1.905 5.11 3480 9.90 3.45 951 2518 4.00 2.100 0.0045 0.0086 1.905 5.11 3480 9.90 0.0045 0.005 0.0086 1.905 5.11 3480 9.90 0.005 0.005 0.0086 0.0086 0.0086 0.0086 0.0086 0.008 0.008 0.008 0.0086 0.0086 0.0086 0.0086 0.0086 0.0086 0.0086 0.0086 0.008 0.0086 0.00 | 97-03 | 1.02 | 0.0 | 0.103 | 67 | 4.07 | 2920 | .68 | ın | ı | | 3,95 | 9.39 | 96.0 | 0.59 | S | 5 |
| 97-04-21 1.02 0.383 0.101 0.680 4.18 2960 3.68 1.52 832 2290 3.96 9.28 0.95 0.58 0.58 0.97 0.58 0.97 0.24 1.02 0.391 0.002 0.765 5.41 3250 4.06 1.52 893 2424 3.98 9.42 0.97 0.58 0.59 0.59 0.59 0.59 0.59 0.59 0.59 0.59 0.59 0.59 0.59 0.59 0.50 0.002 0.765 5.46 3350 4.05 1.42 922 2520 4.01 9.01 0.92 0.95 0.59 0.59 0.703-24 1.02 0.417 0.086 0.800 5.99 3475 4.18 1.46 925 2532 4.02 9.28 0.95 0.59 0.59 0.703-27 1.02 0.417 0.086 0.810 6.34 3375 4.22 1.44 938 2580 4.03 9.20 0.94 0.59 0.59 0.703-27 1.02 0.009 1.495 4.88 3350 7.76 2.82 921 244 4.00 1744 1.79 0.41 0.611 0.70 0.000 1.02 0.000 1.000 0.000 1.000 0.000 1.000 0.000 | 97-01 | 1.02 | 0.385 | 6.106 | 99 | 3.84 | 2825 | 89. | S | . | | 3.96 | 9.78 | 1.00 | 0.58 | ~ | 2 |
| 97-01-24 1.02 0.391 0.097 0.762 3.11 3.20 4.00 1.52 0.95 3.97 0.508 3.57 0.508 3.41 0.508 3.57 0.50 | 97-04 | 1.02 | 0.383 | 0.101 | 68 | 4.18 | 2963 | 899 | in n | ~ - | | 3,96 | 9.28 | 0.95 | C. 58 | \sim $^{\circ}$ | 2; |
| 97-04-24 1.02 0.381 0.087 0.768 5.93 3450 4.05 1.42 922 2520 4.01 9.01 0.92 0.60 97-04-24 1.02 0.417 0.086 0.800 5.99 3475 4.18 1.46 925 2532 4.02 9.28 0.95 0.59 97-01-27 1.02 0.417 0.084 0.810 6.34 3575 4.22 1.44 938 2580 4.03 9.20 0.94 0.59 97-03-37 1.02 0.001 0.001 0.810 6.97 3700 4.12 1.37 94.1 2641 4.06 8.88 0.91 0.61 97-01-30 1.02 0.002 0.002 1.500 4.65 3250 7.86 2.92 901 2418 3.99 18.01 1.85 0.41 97-01-30 1.02 0.410 0.087 1.565 5.23 3460 7.87 2.67 942 2513 4.00 17.16 1.76 0.44 97-01-30 1.02 0.412 0.008 1.585 6.28 3750 7.89 2.67 962 2595 4.03 16.91 1.71 0.44 97-01-30 1.02 0.412 0.008 1.585 6.28 3750 3.00 1.00 0.008 1.585 6.28 3750 3.00 0.008 1.02 0.008 1.000 0.008 1.000 0.008 1.000 0.008 1.000 0.008 1.000 0.008 0.009 0.000 0.008 0.000 0.008 0.000 0.008 0.000 0.008 0.000 0.008 0.000 0.008 0.000 0.008 0.000 0 | 47-03 | 1.02 | 160.0 | 0-0-0 | 2 4 | 5.46 | 3350 | 9 6 | ΛJ | ۰. | | 5.00 | 24.6 | 0.97 | 200 | 1 C | פַּ |
| 97-01-27 1.02 0.417 0.086 C.800 5.99 3475 4.18 1.46 925 2532 4.02 9.28 0.95 0.59 97-01-27 1.02 0.084 0.810 6.34 3575 4.22 1.44 938 2580 4.03 9.20 0.994 0.59 97-01-29 1.02 0.0379 0.081 0.810 6.34 3575 4.22 1.44 938 2580 4.03 9.20 0.994 0.59 97-01-30 1.02 0.0379 0.081 0.899 1.495 3350 4.88 3350 4.82 1.84 1.84 1.85 0. | 97-04 | 1.02 | 0.381 | 0.087 | 26 | 5.93 | 3450 | 0.5 | • 🖈 | . ~ | | 4.01 | 9.01 | 0.92 | 09.0 | 1 ~ | 2 ? |
| 97-03-27 1.02 0.0 0.084 0.810 6.34 3575 4.22 1.44 938 2580 4.03 9.20 0.94 0.559 97-04-27 1.02 0.379 0.081 0.810 6.37 3700 4.12 1.37 947 2641 4.06 8.88 0.91 0.611 0.611 0.612 0.02 0.0379 0.081 0.810 6.87 3700 4.12 1.37 947 2641 4.06 8.88 0.91 0.611 0.611 0.611 0.612 0.02 0.042 0.092 1.500 4.65 3350 7.80 2.92 901 2418 3.99 18.01 1.85 0.41 0.49 0.7-01-36 1.02 0.410 0.087 1.565 5.23 3460 7.87 2.78 2.92 901 2418 3.99 18.01 1.85 0.41 0.44 0.70 0.44 0.70 0.083 1.565 5.81 3530 7.93 2.67 962 2513 4.00 17.16 1.76 0.44 0.70 0.083 1.585 6.28 3750 8.07 2.71 962 2513 4.00 17.16 1.73 0.44 0.70 0.083 1.595 5.75 3530 8.07 2.71 962 2535 4.03 16.68 1.71 0.44 0.70 0.44 0.70 0.088 1.595 5.75 3530 8.07 2.71 960 2539 4.00 21.20 1.71 0.45 0.45 0.098 1.000 0.085 1.900 5.23 3525 9.95 3.41 960 2539 4.00 21.20 2.17 0.45 0.45 0.703-36 1.02 0.008 1.905 5.11 3480 9.90 3.45 951 2518 4.00 21.40 2.20 0.45 0.45 | 97-01 | 1.02 | 0.417 | 0.086 | 90 | 5.99 | 3475 | .18 | | 'n | | 4.02 | 9.28 | 0.95 | 0.59 | \sim | 7.1 |
| 97-04-27 1.62 0.379 0.081 0.810 6.97 3700 4.12 1.37 947 2641 4.06 8.88 0.91 0.611 0.611 97-04-27 1.62 0.379 0.081 0.810 6.97 3700 4.12 1.37 947 2641 4.00 17.44 1.79 0.43 97-01-30 1.02 0.00 0.092 1.500 4.65 3250 7.80 2.92 901 2418 3.99 18.01 1.85 0.44 97-01-30 1.02 0.410 0.087 1.505 5.23 3460 7.87 2.76 942 2513 4.00 17.16 1.76 0.44 97-01-36 1.02 0.482 0.083 1.555 5.81 3530 7.93 2.67 962 2595 4.03 16.91 1.73 0.44 97-01-34 1.02 0.40 0.083 1.595 5.78 3550 8.01 2.41 977 2650 4.05 16.68 1.71 0.44 97-01-34 1.02 0.360 0.085 1.900 5.23 3525 9.95 3.41 960 2539 4.00 21.20 1.70 0.45 97-03-36 1.02 0.00 0.086 1.905 5.11 3480 9.90 3.45 951 2518 4.00 21.40 2.20 0.45 0.45 | 97-03 | 1.02 | 0.0 | 0.084 | 8 | 6.34 | 3575 | •25 | .+ | an. | | 4.03 | 9.20 | 0.94 | 0.59 | 2 | 2 |
| 97-03-30 1.02 0.0 0.090 1.495 4.88 3350 7.76 2.82 921 2464 4.00 17.44 1.79 0.43 97-01-30 1.02 0.422 0.422 0.425 4.88 3350 7.76 2.82 921 2464 4.00 17.44 1.79 0.43 97-01-30 1.02 0.442 0.092 1.500 4.65 3.25 7.60 2.92 901 24.8 3.99 18.01 1.85 0.44 97-01-36 1.02 0.482 0.083 1.565 5.81 3630 7.93 2.67 962 2595 4.03 16.91 1.73 0.44 97-04-33 1.02 0.412 0.080 1.585 6.28 3750 8.01 2.61 977 2650 4.05 16.68 1.71 0.44 97-04-33 1.02 0.360 0.083 1.595 5.75 3630 8.01 2.61 960 2.595 4.04 17.20 1.76 0.45 97-04-34 1.02 0.360 0.088 1.900 5.23 3525 9.95 3.41 960 2539 4.00 21.20 2.77 0.45 97-03-36 1.02 0.360 0.088 1.905 5.11 3480 9.90 3.45 951 2518 4.00 21.40 2.20 0.45 | 97-04 | 1.62 | 0.379 | 0.081 | 81 | 26.9 | 3700 | • | .3 | _ | | 4.06 | 8.88 | 0.91 | 0.61 | \sim | 99 |
| 97-01-30 1.02 0.422 0.092 1.500 4.65 5250 7.80 2.92 901 2418 3.99 18.01 1.85 0.41 97-04-30 1.02 0.4410 0.087 1.505 5.23 3460 7.87 2.76 942 2513 4.00 17.16 1.76 0.44 97-04-36 1.02 0.442 0.080 1.565 5.81 3750 8.01 2.61 977 2650 4.03 16.91 1.71 0.44 97-03-31 1.02 0.0 0.083 1.595 5.75 3630 8.01 2.61 977 2650 4.05 16.68 1.71 0.44 97-03-31 1.02 0.0 0.083 1.595 5.75 3630 8.07 2.71 962 2595 4.04 17.20 1.76 0.45 97-04-36 1.02 0.360 0.085 1.900 5.23 3525 9.95 3.41 960 2539 4.00 21.20 2.17 0.46 97-03-36 1.02 0.0 0.086 1.905 5.11 3480 9.90 3.45 951 2518 4.00 21.40 2.20 0.45 | 97-03 | 1.02 | 0.0 | 060.0 | 6 1 | 4.88 | 3350 | • | 80 | ~ . | | 4.00 | 17.44 | 1.79 | 0.43 | ~ . | 38 |
| 97-01-36 1.02 0.482 0.083 1.565 5.81 3630 7.93 2.67 962 2595 4.03 16.91 1.73 0.44 97-04-33 1.02 0.412 0.080 1.585 6.28 3750 8.01 2.61 977 2650 4.05 16.68 1.71 0.44 97-03-33 1.02 0.0 0.083 1.595 5.75 3630 8.07 2.71 962 2595 4.04 17.20 1.76 0.45 97-04-36 1.02 0.360 0.085 1.900 5.23 3525 9.95 3.41 960 2539 4.00 21.20 2.17 0.46 97-03-36 1.02 0.0 0.086 1.905 5.11 3480 9.90 3.45 951 2518 4.00 21.40 2.20 0.45 | 197-01-30 | 1.02 | 0.422 | 0.092 0.087 | 2 5 | 4.00 000 | 3460 | • | | ٠, | | ٠ د د د د د د د د د د د د د د د د د د د | 18.01 | 1.85 | 44 | | ታ ላ |
| 97-04-33 1.02 0.412 0.080 1.585 6.28 3750 8.01 2.61 977 2650 4.05 16.68 1.71 0.44 97-03-33 1.02 0.0 0.083 1.595 5.75 3630 8.07 2.71 962 2595 4.04 17.20 1.76 0.45 97-04-36 1.02 0.360 0.085 1.900 5.23 3525 9.95 3.41 960 2539 4.00 21.20 2.17 0.46 97-03-36 1.02 0.0 0.086 1.905 5.11 3480 9.90 3.45 951 2518 4.00 21.40 2.20 0.45 | 197-01-36 | 1.02 | 0.482 | 0.083 | 56 | 5.83 | 3630 | | 9 | . ~ | | 4.03 | 16-91 | 1.73 | 44 | - | , (|
| 97-03-33 1.02 0.0 0.083 1.595 5.75 3630 8.07 2.71 962 2595 4.04 17.20 1.76 0.45 97-04-36 1.02 0.360 0.085 1.900 5.23 3525 9.95 3.41 960 2539 4.00 21.20 2.17 0.46 97-03-36 1.02 0.0 0.086 1.905 5.11 3480 9.90 3.45 951 2518 4.00 21.40 2.20 0.45 | 97-04 | 1.02 | 0.412 | 0.080 | 58 | 6.28 | 3750 | | 9 | | | 4.05 | 16.68 | 1.71 | 44 | - | 33. |
| 97-04-36 1.02 0.360 0.085 1.900 5.23 3525 9.95 3.41 960 2539 4.00 21.20 2.17 0.46 97-03-36 1.02 0.0 0.086 1.905 5.11 3480 9.90 3.45 951 2518 4.00 21.40 2.20 0.45 | 97-03 | 1.02 | 0.0 | 0.083 | 5 | 5.75 | 3630 | • | ~ | ~ | | 4.04 | 17.20 | 1.76 | 0.45 | _ | 25 |
| 97-03-36 1.02 0.0 0.086 1.905 5.11 3480 9.90 3.45 951 2518 4.00 21.40 2.20 0.45 | 97-04 | 1.02 | 0.360 | 0.085 | 90 | 5.23 | 3525 | • | 4. | _ | | . 00.4 | 21.20 | 2.17 | 0.46 | _ | 3 |
| | 97-03 | 1.02 | 0.0 | 0.086 | 90 | 5.11 | 3480 | • | 4 | _ | | ** 00 | 21.40 | 2.20 | 0.45 | _ | 23 |

*Free-stream conditions based on frozen flow model.

**Boundary-layer predictions based on nonequilibrium flow-equilibrium catalytic wall conditions.

TABLE V.- HEAT-TRANSFER MEASUREMENTS OBTAINED FOR MODEL I

| | Step, | | | | Hea | ıt-tra | nsfer | rate > | < 10 ⁻⁴ | , watt | s/m² | , at th | ermo | couple | e – | | | |
|----------|-------|------|------|------|------|--------|-------------------|--------|--------------------|-------------------|------|---------|-------|--------|------|------|------|------|
| Run | cm | 1 | 2 | 3 | 4 | 5 | 6 | 7 | 8 | 9 | 10 | 11 | 12 | 13 | 14 | 15 | 16 | 17 |
| | | | | | | (a) p | 0 = 0. | 18 atr | n; T ₍ |) = 28 | 00 K | | | | | | | |
| 180-3-3 | 0.00 | | 2.28 | 2.20 | 2.14 | 2.05 | 2.02 | 1.80 | 1.88 | 1.92 | 1.84 | 1.69 | 1.62 | 1.63 | | 1.59 | 1.57 | 1.57 |
| 180-7-3 | 0.51 | 1.05 | 1.57 | 1.94 | 1.95 | 1.86 | 1.82 | 1.73 | 1.84 | 1.74 | 1.81 | 2.05 | 2.03 | 1.99 | 1.82 | 1.82 | 1.81 | 1.82 |
| 180-1-3 | 1.02 | | 0.65 | 0.77 | 0.79 | 0.80 | 0.90 | 0.91 | 0.89 | 0.91 | 0.82 | 0.90 | 0.92 | 0.92 | | 0.97 | 0.91 | 0.92 |
| | | | | | | (b) p | $_{0} = 0.$ | 20 atr | n; T | ₀ = 32 | 00 K | | | | | | | |
| 180-3-6 | 0.00 | 3.58 | 3.52 | 3.64 | 3.60 | 3.27 | 3.29 | 3.03 | 3.02 | 2.96 | 3.04 | 2.72 | 2.84 | 2.76 | | 2.68 | 2,51 | 2.46 |
| 180-5-6 | 0.16 | 2.85 | 3.00 | 3.19 | 3.11 | 2.85 | 2.62 | 2.56 | 2.51 | 2.44 | 2.57 | 3.25 | 3.35 | 3.18 | | 2.05 | 2.23 | 2.39 |
| 180-7-6 | 0.51 | 1.63 | 2.31 | 2.77 | 2.84 | 2.73 | 2.84 | 2.84 | 2.87 | 2.85 | 2.81 | | 3.05 | 2.96 | 2.71 | 2.73 | 2.68 | 2.69 |
| 180-1-6 | 1.02 | 0.38 | 0.85 | 1.25 | 1.48 | 1.59 | 1.70 | 1.73 | 1.70 | 1.73 | 1.83 | 1.84 | 1.84 | 1.70 | | 1.70 | 1.67 | 1.62 |
| | | | | | | (c) p | ₀ = 0. | 22 atr | n; T |) = 34 | 00 K | | | | | | | |
| 180-3-9 | 0.00 | 4.42 | 4.29 | 4.53 | 4.33 | 4.01 | 4.02 | 3.51 | 3.69 | 3.68 | 3.75 | 3.41 | 3.26 | 3.37 | | 3.33 | 3.02 | 3.02 |
| 180-5-9 | 0.16 | 3.55 | 3.91 | 4.21 | 3.99 | 3.72 | 3.47 | 3.17 | 3.14 | 3.09 | 3.27 | 3.76 | 3.93 | 3.76 | | 2.82 | 2.84 | 2.85 |
| 180-7-9 | 0.51 | 2.13 | 3.02 | 3.51 | 3.70 | 3.52 | 3.58 | 3.64 | 3.64 | 3.59 | 3.53 | 3.63 | 3.64 | 3.59 | 3.39 | 3.42 | 3.34 | 3.30 |
| 180-1-9 | 1.02 | 0.57 | 1.35 | 1.89 | 2.15 | 2.23 | 2.45 | 2.53 | 2.61 | 2.62 | 2.72 | 2.70 | 2.72 | 2.73 | | 2.43 | 2.40 | 2.39 |
| , | | | • | | | (d) p | $_{0} = 0.$ | 24 atr | n; T |) = 38 | 00 K | | | | | | | |
| 180-3-12 | 0.00 | 5.35 | 5.13 | 5.33 | 5.17 | 4.67 | 4.60 | 4.24 | 4.34 | 4.38 | 4.36 | 4.12 | 3.86 | 3.96 | | 3.93 | 3.59 | 3.56 |
| 180-5-12 | 0.16 | 4.57 | 4.61 | 5.01 | 4.84 | 4.65 | 4.28 | 3.94 | 3.96 | 3.87 | 3.97 | 4.59 | 4.61 | 4.50 | | 3.76 | 3.61 | 3.51 |
| 180-7-12 | 0.51 | 2.44 | 3.52 | 4.15 | 4.33 | 4.80 | | 4.21 | 4.33 | 4.16 | 4.36 | 4.26 | 4.21 | 4.10 | 3.97 | 4.02 | 3.97 | 3.88 |
| 180-1-12 | 1.02 | 0.80 | 1.58 | 2.17 | 2.61 | 2.63 | 2.90 | 3.06 | 3.14 | 3.19 | 3.22 | 3.25 | 3.23ي | 3.25 | | 3.14 | | 3.02 |
| | | | | | | (e) p | 0 = 0. | 25 atr | n; . T | o = 45 | 00 K | | | | | | | |
| 180-3-15 | 0.00 | 6.80 | 6.58 | 6.69 | 6.58 | 6.18 | 6.13 | 5.75 | 5.75 | 5.64 | 5.56 | 5.35 | 5.10 | 5.10 | | 4.93 | 4.67 | 4.64 |
| 180-5-15 | 0.16 | 5.56 | 5.95 | 6.23 | 6.09 | 5.69 | 5.46 | 5.35 | 5.30 | 5.33 | 5.35 | 5.28 | 5.28 | 5.25 | | 4.65 | 4.44 | 4.36 |
| 180-7-15 | 0.51 | 2.95 | 4.38 | 5.16 | 5.35 | 5.13 | 5.22 | 5.18 | 5.32 | 5.28 | 5.30 | 5.33 | 5.31 | 5.20 | 4.98 | 5.10 | 4.95 | 4.92 |
| 180-1-15 | 1.02 | 1.06 | 2.05 | 2.82 | 3.36 | 3.44 | 3.72 | 3.88 | | 3.96 | 3.97 | 3.97 | 3.94 | 3.99 | | 3.79 | 3.68 | 3.64 |
| | | | | | | (f) p | 0 = 0. | 27 atn | n; T ₍ | = 50 | 00 K | | | | | | | |
| 180-3-18 | 0.00 | 8.01 | 7.65 | 7.70 | 7.62 | 7.16 | 7.23 | 6.78 | 6.80 | 6.69 | 6.50 | 6.29 | 6.15 | 5.92 | | 5.74 | 5.47 | 5.35 |
| 180-5-18 | 0.16 | 6.61 | 6.96 | 7.22 | 7.18 | 6.66 | 6.72 | 6.32 | 6.43 | 6.37 | 6.26 | 6.24 | 6.21 | 6.09 | | 5.60 | 5.47 | 5.30 |
| 180-7-18 | 0.51 | 3.59 | 5.15 | 6.09 | 6.32 | 6.10 | 6,10 | 6.12 | 6.22 | 6.08 | 6.06 | 6.26 | 6.21 | 6.07 | 5.81 | 5.87 | 5.67 | 5.68 |
| 180-1-18 | 1.02 | 1.31 | 2.52 | 3.48 | 4.09 | 4.21 | 4.65 | 4.78 | 4.85 | 4.89 | 4.94 | 1.14 | 4.90 | 4.87 | | 4.66 | 4.53 | 4.44 |

TABLE V.- HEAT-TRANSFER MEASUREMENTS OBTAINED FOR MODEL I - Continued

| | Step, | | | | Heat- | trans | fer ra | ate × 1 | 0-4, | watts, | / _{m², a} | at ther | mocc | uple | <u></u> | | | |
|----------|-------|---------------|------|------|-------|-------|-------------------|---------|------|--------|-------------------------------|---------|------|------|---------|------|------|------|
| Run | cm | 1 | 2 | 3 | 4 | 5 | 6 | 7 | 8 | 9 | 10 | 11 | 12 | 13 | 14 | 15 | 16 | 17 |
| | | | | | | (g) p | 0 = 0. | 28 atı | n; T | 0 = 52 | 00 K | | | | | | | |
| 180-3-21 | 0.00 | 9.17 | 8.91 | 8.84 | 8.63 | 8.05 | 8.09 | 7.86 | 7.78 | 7.57 | 7.46 | 7.10 | 6.95 | 6.88 | | 6.70 | 6.38 | 6.15 |
| 180-5-21 | 0.16 | 7.46 | 8.03 | | | | I | l | 7.46 | 1 | 7.13 | | 7.27 | 7.08 | | 6.61 | 6.46 | 6.26 |
| 180-7-21 | 0.51 | 4.10 | 5.77 | 6.92 | 7.00 | 6.82 | 6.84 | 6.87 | 7.05 | 6.83 | 6.83 | 7.06 | 7.06 | 6.92 | 6.54 | 6.47 | 6.36 | 6.38 |
| 180-1-21 | 1.02 | 1.61 | 3.07 | 4.23 | 4.98 | 5.24 | 5.73 | 5.81 | 5.84 | 5.92 | 5.98 | 5.92 | 5.87 | 5.81 | | 5.69 | 5.58 | 5.42 |
| | | | | | | (h) p | 0 = 0 | .36 atı | n; T | 0 = 30 | 00 K | | | | | · | | |
| 180-3-24 | 0.00 | 3.84 | 3.52 | 3.54 | 3.47 | 3.25 | 3.28 | 2.96 | 3.05 | 2.96 | 2.96 | 2.85 | 2.62 | 2.68 | | 2.71 | 2.51 | 2.45 |
| 180-6-3 | 0.16 | | 2.96 | 1 1 | | | | 2.62 | l | l : | | i | | 2.87 | | 2.41 | 2.33 | 2.24 |
| 180-7-24 | 0.51 | 1.40 | 2.11 | 2.64 | 2.76 | 2.73 | 2.73 | 2.73 | 2.82 | 2.76 | 2.77 | 2.93 | 3.02 | 2.82 | 2.85 | 2.82 | 2.73 | 2.71 |
| 180-1-24 | 1.02 | 0.46 | 0.93 | 1.37 | 1.71 | 1.80 | 1.99 | 2.04 | 2.05 | 2.10 | 2.05 | 2.12 | 2.11 | 2.10 | | 2.13 | 2.05 | 2.04 |
| | | | | | | (i) p | 0 = 0. | 38 atn | n; T |) = 32 | 00 K | | | | | | | |
| 180-3-27 | 0.00 | 4.56 | 4.26 | 4.33 | 4.27 | 3.87 | 3.94 | 3.47 | 3.66 | 3.62 | 3.55 | 3.30 | 3.11 | 3.22 | | 3.25 | 3.00 | 2.94 |
| 180-6-6 | 0.16 | 3.51 | 3.70 | 4.05 | 3.95 | 3.64 | 3.53 | 3.25 | 3.30 | 3.19 | 3.30 | 3.59 | 3.68 | 3.53 | | 3.14 | 2.96 | 2.85 |
| 180-7-27 | 0.51 | 1.79 | 2.62 | 3.19 | 3.34 | 3.26 | 3.29 | 3.25 | 3.38 | 3.37 | 3.36 | 3.59 | 3.62 | 3.53 | 3.33 | 3.35 | 3.22 | 3.20 |
| 180-1-27 | 1.02 | | 1.28 | 1.82 | 2.16 | 2.24 | 2.41 | 2.57 | 2.61 | 2.66 | 2.61 | 2.62 | 2.61 | 2.70 | | 2.71 | | 2.61 |
| | | · · · · · · · | | | | (j) p | 0 = 0. | 40 atn | n; T | 0 = 34 | 00 K | | | | | | | |
| 180-3-30 | 0.00 | 5.64 | 5.20 | 5.27 | 5.24 | 4.73 | 4.81 | 4.30 | 4.38 | 4.34 | 4.37 | 4.07 | 3.85 | 4.03 | | 3.91 | 3.59 | 3.63 |
| 180-6-9 | 0.16 | 4.10 | 4.36 | 4.82 | 4.70 | 4.33 | 3.99 | 3.84 | 3.85 | 3.82 | 3.89 | 4.27 | 4.35 | 4.16 | | 3.64 | 3.53 | 3.45 |
| 180-7-30 | 0.51 | 2.06 | 3.07 | 3.80 | 3.96 | 3.85 | 3.91 | 4.04 | 4.10 | 4.01 | 4.02 | 4.20 | 4.28 | 4.16 | 3.93 | 4.02 | 3.87 | 3.89 |
| 180-1-30 | 1.02 | | 1.49 | 2.06 | 2.45 | 2.53 | 2.88 | 3.03 | 3.07 | 3.05 | 3.05 | 3.06 | 3.09 | 3.09 | | 3.07 | 2.96 | 2.93 |
| | | | | | | (k) p | 0 = 0 | .44 atı | n; T | 0 = 38 | 800 K | | | | | | | |
| 180-3-33 | 0.00 | 7.40 | 6.95 | 6.95 | 6.86 | 6.32 | 6.18 | 6.07 | | 5.84 | 5.79 | 5.39 | 5.17 | 5.36 | | 5.17 | 4.81 | 4.87 |
| 180-6-12 | 0.16 | 5.80 | 6.04 | 6.59 | 6.55 | 5.98 | 5.68 | 5.30 | 5.35 | 5.38 | 5.35 | 5.81 | 5.92 | 5.73 | | 5.01 | 4.81 | 4.67 |
| 180-7-33 | 0.51 | 2.64 | 4.04 | 4.93 | 5.15 | 5.07 | 5.17 | 5.13 | 5.23 | 5.14 | 5.17 | 5.26 | 5.27 | 5.22 | 4.93 | 4.94 | 4.76 | 4.78 |
| | | | | | · | (l) p | ₀ = 0. | 47 atr | n; T | 0 = 44 | 00 K | | | | | | | |
| 180-3-36 | 0.00 | 9.00 | 8.52 | 8.60 | 8.26 | 7.66 | 7.74 | 7.23 | 7.27 | 7.12 | 6.99 | 6.64 | 6.40 | 6.47 | | 6.37 | 5.92 | 5.81 |
| 180-6-15 | 0.16 | 6.83 | 7.40 | l 1 | | 7.27 | 7.06 | | 1 | 6.66 | | | | 6.72 | | 6.15 | 5.91 | 1 } |
| 180-7-36 | 0.51 | 3.30 | 5.24 | l 1 | | 6.32 | 6.56 | | | 6.58 | | | 6.45 | 6.39 | 6.13 | 5.75 | 6.09 | |
| 180-2-3 | 1.02 | 1.10 | 2.12 | 3.05 | 3.69 | 3.87 | 4.27 | 4.37 | 4.46 | 4.54 | 4.73 | 4.78 | 4.70 | 4.67 | | 4.67 | 4.44 | 4.46 |

TABLE V.- HEAT-TRANSFER MEASUREMENTS OBTAINED FOR MODEL I - Continued

| | Step, Heat-transfer rate × 10 ⁻⁴ , watts/m ² , at thermocouple - | | | | | | | | | | | | | | | | | |
|--|--|------|------|------|------|-------|-------------|--------|-------------------|---------|------|---------|------|------|------|--|------|--------------|
| Run | cm | 1 | 2 | 3 | 4 | 5 | 6 | . 7 | 8 | 9 | 10 | 11 | 12 | 13 | 14 | 15 | 16 | 17 |
| | | | · | | (| m) p | $_{0} = 0.$ | 50 atn | n; T ₍ | = 47 | 00 K | | | | | . | l —. | |
| 180-3-39 | 0.00 | 10.5 | 10.0 | 9.79 | 9.57 | 9.34 | 9.09 | 8.61 | 8.62 | 8.43 | 8.31 | 7.95 | 7.63 | 7.52 | | 7.40 | 7.04 | 6.97 |
| 180-6-18 | 0.16 | 8.53 | 9.13 | 9,62 | 9.64 | 9.05 | 9.17 | 8.83 | 8.83 | 8.66 | 8.60 | 8.47 | 8.43 | 8.20 | | 7.74 | 7.52 | 7.40 |
| 180-7-39 | 0.51 | 3.99 | 6.21 | 7.38 | 7.68 | 7.40 | 7.70 | 7.63 | 7.74 | 7.65 | | 7.63 | 7.63 | 7.54 | 7.18 | 7.23 | 7.07 | 7.11 |
| 180-2-6 | 1.02 | 1.37 | 2.78 | 4.10 | 4.89 | 5.09 | 5.63 | 5.67 | 5.88 | 5.84 | 6.04 | 6.00 | 5.93 | 5.90 | | 5.85 | 5.67 | 5.57 |
| (n) $p_0 = 0.64 \text{ atm}; T_0 = 2800 \text{ K}$ | | | | | | | | | | | | | | | | | | |
| 180-4-3 | 0.00 | 3.47 | 3.30 | 3.22 | 3.15 | 2.93 | 2.95 | 2.77 | 2.73 | 2.62 | 2.53 | 2.43 | 2.38 | 2.36 | | 2.29 | 2.28 | 2.20 |
| 180-6-21 | 0.16 | 2.45 | 2.68 | 2.85 | 2.88 | 2.73 | 2.72 | 2.62 | 2.56 | 2.53 | 2.56 | 2.50 | 2.61 | 2.64 | | 2.60 | 2.72 | 2.96 |
| 180-7-42 | 0.51 | 0.93 | 1.78 | 2.27 | 2.39 | 2.39 | 2.45 | 2.49 | 2.51 | 2.53 | 2.56 | 2.65 | 2.64 | | 2.51 | 2.52 | 2.52 | 2.62 |
| 180-2-9 | 1.02 | 0.38 | 0.69 | 0.97 | 1.25 | 1.48 | 1.64 | 1.72 | 1.80 | 1.81 | 1.86 | 1.91 | 1.90 | 1.87 | | 1.92 | 1.83 | 1.82 |
| (o) $p_0 = 0.68 \text{ atm}; T_0 = 3000 \text{ K}$ | | | | | | | | | | | | | | | | | | |
| 180-4-6 | 0.00 | 4.36 | 4.18 | 4.11 | 4.05 | 3.77 | 3.80 | 3.51 | 3.47 | 3.44 | 3.42 | 3.21 | 3.15 | 3.14 | | 3.07 | 2.96 | 2.94 |
| 180-6-24 | 0.16 | 3.19 | 3.64 | 3.87 | 3.82 | 3.59 | 3.52 | 3.37 | 3.44 | 3.42 | 3.36 | 3.49 | 3.64 | 3.55 | | 3.15 | 3.04 | 3.05 |
| 180-7-45 | 0.51 | 1.39 | 2.39 | 3.04 | 3.21 | 3.12 | 3.22 | 3.30 | 3.42 | 3.33 | 3.42 | 3.62 | 3.64 | 3.53 | 3.33 | 3.31 | 3.30 | 3.37 |
| 180-2-12 | 1.02 | 0.49 | 0.84 | 1.37 | 1.82 | 1.97 | 2.20 | 2.28 | 2.31 | 2.37 | 2.48 | 2.52 | 2.53 | 2.48 | | 2.52 | 2.39 | 2.37 |
| | | | | | (| (p) p | 0 = 0. | 71 atn | n; T ₍ | = 320 | 00 K | | | | | | | |
| 180-4-9 | 0.00 | 5.42 | 5.24 | 5.39 | 5.20 | 4.73 | 4.81 | 4.45 | 4.48 | 4.40 | 4.35 | 4.29 | 4.02 | 4.08 | | 3.96 | 3.79 | 4.35 |
| 180-6-27 | 0.16 | 3.80 | 4.27 | 4.67 | 4.62 | 4.35 | 4.30 | 4.10 | 4.15 | 4.12 | 4.18 | 4.30 | 4.30 | 4.25 | | 3.77 | 3.68 | 3.52 |
| 180-7-48 | 0.51 | 1.71 | 2.90 | 3.68 | 3.87 | 3.79 | 3.92 | 3.96 | 4.00 | 4.01 | 4.02 | 4.10 | 4.15 | 3.97 | 3.99 | 3.97 | | 3.99 |
| 180-2-15 | 1.02 | 0.36 | 0.71 | 1.09 | 1.39 | 1.59 | 1.72 | 1.72 | 1.74 | 1.72 | 1.72 | 1.72 | 1.72 | 1.69 | | 1.67 | 1.59 | 1.58 |
| | | | | | | (q) p | $_{0}=0$. | 76 atn | n; T ₍ | = 35 | 00 K | | | | | | | |
| 180-4-12 | 0.00 | 7.23 | 6.94 | 6.97 | 6.83 | 6.26 | 6.36 | 5.82 | 5.98 | 5.82 | 5.74 | 5.47 | 5.24 | 5.30 | | 5.16 | 4.86 | 4.84 |
| 180-6-30 | 0.16 | 5.23 | 5.80 | | | i | | 1 | | 5.46 | | 5.84 | | 5.69 | | 5.01 | 4.78 | 4.73 |
| 180-7-51 | 0.51 | 2.25 | 3.76 | 4.78 | 5.11 | 4.95 | 5.07 | 5.13 | 5.24 | 5.18 | 5.24 | 5.47 | 5.58 | 5.41 | 5.01 | 4.98 | 4.90 | 4.98 |
| 180-2-18 | 1.02 | 0.56 | 1.37 | 1.84 | 2.36 | 2.60 | 2.84 | 2.82 | 1 | l | | 1 | | 2.58 | | 2.51 | 2.28 | 2.16 |
| | | · | | | | (r) p | $_{0}=0.$ | 80 atn | 1; T ₍ | = 370 | 00 K | | -,- | · | | <u>. </u> | | |
| 180-4-15 | 0.00 | | 8.49 | 8.43 | | | | 7.03 | ` | <u></u> | | 6.59 | 6.33 | 6.40 | | 6.21 | 5.87 | 5.81 |
| 180-6-33 | 0.16 | 6.33 | 7.10 | | | 1 | | 6.51 | l | 1 | | 6.81 | l | 6.59 | | 6.03 | 5.72 | 5.64 |
| 180-7-54 | 0.51 | 2.99 | 5.07 | 6.30 | 6.60 | 6.47 | 6.58 | 6.82 | 6.89 | 6.85 | 6.88 | 7.11 | 7.04 | 6.84 | 6.56 | 6.57 | 6.53 | 6.53 |

TABLE V.- HEAT-TRANSFER MEASUREMENTS OBTAINED FOR MODEL I - Concluded

| 7. | Step, | | | | Hea | ıt-trai | ısfer | rate > | 10-4 | , watt | s/m ² | , at th | ermo | couple | · - | | | |
|--|--|------|------|------|------|---------|--------|--------|------|--------|------------------|---------|------|--------|-------------|------|------|------|
| Run | cm ` | 1 | 2 | 3 | 4 | 5 | 6 | 7 | 8 | 9 | 10 | 11 | 12 | 13 | 14 | 15 | 16 | 17 |
| | (s) $p_0 = 0.85 \text{ atm}$; $T_0 = 4000 \text{ K}$ $180-4-18 \ 0.00 \ 10.9 \ 10.4 \ 10.4 \ 9.97 \ 9.28 \ 9.40 \ 9.04 \ 8.71 \ 8.66 \ 8.46 \ 7.97 \ 7.74 \ 7.78 \ \ 7.65 \ 7.12 \ 6.92$ | | | | | | | | | | | | | | | | | |
| 180-4-18 | 0.00 | 10.9 | 10.4 | 10.4 | 9.97 | 9.28 | 9.40 | 9.04 | 8.71 | 8.66 | 8.46 | 7.97 | 7.74 | 7.78 | | 7.65 | 7.12 | 6.92 |
| 180-6-36 | 0.16 | 8.07 | 8.94 | 9.42 | 9.33 | 8.76 | 8.66 | 8.43 | 8.37 | 8.31 | 8.29 | 8.23 | 8.23 | 8.14 | | 7.54 | 7.35 | 7.15 |
| 180-7-57 | 0.51 | 3.54 | 6.04 | 7.52 | 7.86 | 7.72 | 7.93 | 8.01 | 8.12 | 8.07 | 8.09 | 8.36 | 8.34 | 8.22 | 7.74 | 7.77 | 7.57 | 10.1 |
| 180-2-24 | 1.02 | 0.99 | 2.07 | 3.35 | 4.16 | 4.51 | 5.05 | 5.01 | 5.20 | 5.12 | 5.30 | 5.07 | 4.85 | 4.60 | | 4.46 | 3.84 | 3.58 |
| (t) $p_0 = 1.36 \text{ atm}; T_0 = 3200 \text{ K}$ | | | | | | | | | | | | | | | | | | |
| 180-11-3 | 0.00 | 6.61 | 6.31 | 6.18 | 5.83 | 5.56 | 5.65 | 5.35 | 5.36 | | 5.24 | 5.41 | 5.58 | | 5.34 | 5.33 | 5.31 | 6.43 |
| 180-10-3 | 0.16 | 4.44 | 5.03 | 5.38 | 5.42 | 5.24 | 5.16 | 5.09 | 5.01 | 4.90 | 5.01 | 5.09 | 5.10 | 5.01 | 4.81 | 4.87 | 4.87 | 4.76 |
| 180-9-3 | 0.51 | 1.54 | 2.85 | 3.87 | 4.17 | 4.13 | 4.33 | 4.36 | 4.37 | 4.38 | 4.51 | 4.78 | 4.77 | 4.67 | 4.40 | 4.44 | 4.40 | 4.67 |
| 180-8-3 | 1.02 | 0.34 | 0.67 | 1.20 | 1.69 | 1.96 | 2.58 | 2.97 | 3.07 | 3.11 | 3.22 | 3.42 | 3.54 | 3.55 | 3.39 | 3.37 | 3.36 | 4.16 |
| | | | | | | (u) p | 0 = 1. | 42 atr | n; T |) = 35 | 00 K | | | | · · · · · · | | | 1 |
| 180-11-6 | 0.00 | 8.09 | 7.54 | 7.63 | 7.40 | 6.87 | 6.96 | 6.59 | 6.56 | 6.38 | 6.55 | 6.58 | 6.54 | 6.38 | 6.15 | 6.30 | 6.25 | 7.54 |
| 180-10-6 | 0.16 | 5.34 | 6.30 | 6.67 | 6.72 | 6.48 | 6.45 | 6.32 | 6.15 | 5.97 | 6.15 | 6.42 | 6.38 | 6.15 | 5.92 | 6.04 | 6.04 | |
| 180-9-6 | 0.51 | 2.05 | 3.76 | 4.84 | 5.18 | 5.23 | 5.31 | 5.50 | 5.24 | 5.09 | 5.37 | 5.69 | 5.61 | 5.44 | 5.18 | 5.06 | 4.99 | 5.56 |
| 180-8-6 | 1.02 | 0.48 | 0.85 | 1.71 | 2.33 | 2.65 | 3.20 | 3.42 | 3.53 | 3.60 | 3.79 | 4.16 | 4.20 | 4.13 | 3.99 | 4.07 | 4.23 | 5.12 |
| | | | | | | (v) p | 0 = 1. | 50 atr | n; T |) = 38 | 00 K | | | | | | | |
| 180-11-9 | 0.00 | 9.62 | 9.26 | 9.41 | 9.05 | 8.43 | 8.54 | 8.02 | 8.05 | 7.73 | 7.80 | 7.85 | 7.85 | 7.68 | 7.32 | 7.46 | 7.37 | 8.83 |
| 180-10-9 | 0.16 | 6.96 | 7.74 | 8.15 | 8.19 | 7.82 | 7.72 | 7.63 | 7.35 | 7.18 | 7.35 | 7.61 | 7.45 | 7.35 | 6.83 | 6.88 | 6.95 | 8.09 |
| 180-9-9 | 0.51 | 2.45 | 4.19 | 5.92 | 6.32 | 6.39 | 6.43 | 6.49 | 6.39 | 6.30 | 6.61 | 7.04 | 7.03 | 6.81 | 6.39 | 6.39 | 6.48 | 7.47 |
| 180~8-9 | 1.02 | 0.57 | 1.14 | 2.20 | 3.01 | 3.66 | 4.11 | 4.33 | 4.38 | 4.50 | 4.90 | 5.23 | 5.24 | 5.12 | 4.84 | 4.92 | 5.26 | 6.21 |

TABLE VI.- HEAT-TRANSFER MEASUREMENTS OBTAINED FOR MODEL II

| | Step, | Suction rate, | | I | -teat−t | ransf | er rat | e × 10 | -4, w | atts/r | n ² , at | ther | nocou | ple - | - | |
|---|--|---------------|------------------|------------------|---------|--------|-------------------|--------|---------|--------|---------------------|------|-------|-------|------|------|
| Run | cm | w | 1 | 2 | 3 | 4 | 5 | 6 | 7 | 8 | 9 | 10 | 11 | 12 | 13 | 14 |
| | (a) $p_0 = 0.22 \text{ to } 0.23 \text{ atm}; T_0 = 3700 \text{ K}$ | | | | | | | | | | | | | | | |
| 188-5-3 | 1.02 | 0.25 | 0.82 | 1.56 | 2.11 | 2.58 | 2.91 | 3.08 | 3.22 | 3.24 | 3.31 | 3.31 | 3.25 | 3.14 | 3.15 | 3.10 |
| 190-8-3 | 1.02 | 0.00 | 0.89 | 1.63 | 2.27 | 2.62 | 3.10 | 3.24 | 3.37 | 3.37 | 3.48 | 3.22 | 3.27 | 3.33 | 3.34 | 3.30 |
| 190-10-3 | 0.00 | 0.00 | 6.07 | 5.85 | 5.64 | 5.43 | 5.23 | 4.94 | 4.85 | 4.67 | 4.49 | 4.34 | 4.16 | 4.03 | 3.86 | 3.74 |
| | (b) $p_0 = 0.35 \text{ to } 0.38 \text{ atm}; T_0 = 3200 \text{ K}$ | | | | | | | | | | | | | | | |
| 190-7-12 | 1.02 | 0.22 | 0.61 | 1.17 | 1.56 | 1.85 | 2.13 | 2.46 | 2.49 | 2.39 | 2.51 | 2.39 | 2.40 | 2.41 | 2.54 | 2.51 |
| 190-8-12 | 1.02 | 0.00 | 0.39 | 0.73 | 1.23 | 1.62 | 2.02 | 2.32 | 2.31 | 2.29 | 2.50 | 2.17 | 2.21 | 2.39 | 2.42 | 2.34 |
| 1,90-10-12 | 0.00 | 0.00 | 4.70 | 4.51 | 4.37 | 4.20 | 4.07 | 3.94 | 3.82 | 3.67 | 3.56 | 3.43 | 3.34 | 3.23 | 3.13 | 3.06 |
| (c) $p_0 = 0.40 \text{ to } 0.43 \text{ atm}$; $T_0 = 3600 \text{ to } 3900 \text{ K}$ | | | | | | | | | | | | | | | | |
| 190-7-15 | 1.02 | 0.28 | 1.07 | 1.84 | 2.58 | 3.11 | 3.77 | 4.14 | 4.13 | 3.97 | 4.32 | 4.06 | 4.04 | 4.17 | 4.24 | 4.01 |
| 190-8-15 | 1.02 | 0.00 | 0.71 | 1.27 | 2.01 | 2.76 | 3.34 | 3.83 | 3.85 | 3.74 | 4.16 | 3.62 | 3.71 | 3.97 | 4.02 | 3.84 |
| 190-10-15 | 0.00 | 0.00 | 7.79 | 7.51 | 7.27 | 7.02 | 6.78 | 6.54 | 6.33 | 6.08 | 5.85 | 5.62 | 5.42 | 5.22 | 5.04 | 4.83 |
| | | | (d) | p ₀ = | = 0.43 | atm; | T ₀ = | 3900 | K | | | | | | | |
| 190-3-6 | 1.02 | 0.40 | 1.92 | 2.84 | 3.63 | 4.19 | 4.53 | 4.66 | 4.74 | 4.76 | 4.89 | 4.91 | 4.96 | 4.84 | 4.87 | 5.09 |
| 190-8-15 | 1.02 | 0.00 | 0.71 | 1.27 | 2.01 | 2.76 | 3.34 | 3.83 | 3.85 | 3.74 | 4.16 | 3.62 | 3.71 | 3.97 | 4.02 | 3.84 |
| 190-10-15 | 0.00 | 0.00 | 7.79 | 7.51 | 7.27 | 7.02 | 6.78 | 6.54 | 6.33 | 6.08 | 5.85 | 5.62 | 5.42 | 5.22 | 5.04 | 4.83 |
| | | (e) | р ₀ = | = 0.43 | to 0. | 46 atn | ı; T ₀ | = 430 | 00 to 4 | 1400 F | | _ | | , | | |
| 190-7-18 | 1.02 | 0.26 | 1.48 | 2.46 | 3.51 | 4.27 | 4.86 | 5.12 | 5.28 | 5.22 | 5.31 | 5.21 | 5.22 | 5.24 | 5.22 | 5.18 |
| 190-8-18 | 1.02 | 0.00 | 0.86 | 1.85 | 2.93 | 3.79 | 4.39 | 4.73 | 4.90 | 4.87 | 5.09 | 4.93 | 5.06 | 5.09 | 5.06 | 5.13 |
| 190-10-18 | 0.00 | 0.00 | 9.83 | 9.54 | 9.26 | 8.95 | 8.70 | 8.42 | 8.18 | 7.96 | 7.70 | 7.48 | 7.23 | 7.01 | 6.77 | 6.57 |
| | | (f) | p ₀ = | 0.62 | to 0.6 | 39 atm | ; T ₀ | = 290 | 0 to 3 | 000 K | | | | | | |
| 188-6-15 | 1.02 | 0.34 | 0.72 | 1.23 | 1.66 | 1.97 | 2.18 | 2.21 | 2.31 | 2.17 | 2.18 | 2.07 | 2.16 | 2.17 | 2.33 | 2.38 |
| 190-8-21 | 1.02 | 0.00 | 0.36 | 0.65 | 0.78 | 0.94 | 1.34 | 1.47 | 1.59 | 1.64 | 1.77 | 1.64 | 1.65 | 1.66 | 1.70 | 1.61 |
| 190-10-21 | 0.00 | 0.00 | 4.52 | 4.34 | 4.16 | 4.82 | 3.90 | 3.77 | 3.64 | 3.53 | 3.41 | 3.30 | 3.20 | 3.11 | 3.01 | 2.91 |

TABLE VI.- HEAT-TRANSFER MEASUREMENTS OBTAINED FOR MODEL II - Continued

| ו מווא | Step, | Step Suction rate Heat-transfer rate × 10-4, watts/m ² , at thermocouple - | | | | | | | | | | | | | | |
|--|-------|---|------|--------------------|--------------|--------|------|--------------------|--------|------|------|------|------|------|------|------|
| ' | | Suction rate, w | | | - | T | | Τ | | 1 | | | i | | | 1 |
| | cm | | 1 | 2 | 3 | 4 | 5 | 6 | 7 | 8 | 9 | 10 | 11 | 12 | 13 | 14 |
| (g) $p_0 = 0.65 \text{ to } 0.69 \text{ atm}$; $T_0 = 2900 \text{ to } 3000 \text{ K}$ 190-7-21 1.02 0.35 0.70 1.10 1.54 1.87 2.15 2.34 2.32 2.20 2.40 2.14 2.19 2.17 2.08 2.96 2.14 2.18 2 | | | | | | | | | | | | | | | | |
| 190-7-21 1 | 1.02 | 0.35 | 0.70 | 1.10 | 1.54 | 1.87 | 2.15 | 2.34 | 2.32 | 2.20 | 2.40 | 2.14 | 2.19 | 2.17 | 2.08 | 2.96 |
| 190-8-21 | 1.02 | 0.00 | 0.36 | 0.65 | 0.78 | 0.94 | 1.34 | 1.47 | 1.59 | 1.64 | 1.77 | 1.64 | 1.65 | 1.66 | 1.70 | 1.61 |
| 190-10-21 | 0.00 | 0.00 | 4.52 | 4.34 | 4.16 | 4.82 | 3.90 | 3.77 | 3.64 | 3.53 | 3.41 | 3.30 | 3.20 | 3.11 | 3.01 | 2.91 |
| (h) $p_0 = 0.71 \text{ to } 0.77 \text{ atm}; T_0 = 3400 \text{ K}$ | | | | | | | | | | | | | | | | |
| 190-2-18 1 | 1.02 | 0.41 | 1.38 | 2.23 | 2.81 | 2.73 | 3.22 | 3.36 | 3.45 | 3.43 | 3.57 | 3.43 | 3.51 | 3.37 | 3.37 | 3.51 |
| 190-7-24 1 | 1.02 | 0.36 | 1.01 | 1.63 | 2.27 | 2.73 | 3.12 | 3.55 | 3.56 | 3.31 | 3.58 | 3.32 | 3.22 | 3.36 | 3.24 | 3.10 |
| 190-8-24 | 1.02 | 0.00 | 0.31 | 0.69 | 1.14 | 1.60 | 2.20 | 2.46 | 2.60 | 2.63 | 2.87 | 2.62 | 2.71 | 2.67 | 2.68 | 2.59 |
| 190-10-24 | 0.00 | 0.00 | 7.24 | 6.94 | 6.69 | 6.45 | 6.24 | 6.02 | 5.81 | 5.63 | 5.43 | 5.26 | 5.06 | 4.89 | 4.72 | 4.58 |
| (i) $p_0 = 0.75 \text{ to } 0.81 \text{ atm}; T_0 = 3700 \text{ K}$ | | | | | | | | | | | | | | | | |
| 190-7-27 1 | 1.02 | 0.40 | 1.35 | 2.11 | 2.89 | 3.50 | 4.13 | 4.58 | 4.55 | 4.38 | 4.59 | 4.16 | 4.18 | 4.35 | 4.33 | 4.05 |
| 190-8-27 1 | 1.02 | 0.00 | 0.42 | 0.85 | 1.57 | 2.21 | 2.91 | 3.35 | 3.49 | 3.55 | 3.81 | 3.43 | 3.54 | 3.57 | 3.58 | 3.45 |
| 190-10-27 0 | 0.00 | 0.00 | 8.90 | 8.52 | 8.22 | 7.91 | 7.63 | 7.37 | 7.07 | 6.82 | 6.55 | 6.34 | 6.11 | 5.85 | 5.63 | 5.42 |
| | | | (j) | p ₀ = 0 | 0.80 to | 0.87 | atm; | T ₀ = | = 3900 | K | | | | | | |
| 188-2-9 1 | 1.02 | 0.44 | 2.07 | 3.58 | 4.89 | 5.59 | 6.14 | 6.36 | 6.53 | 6.31 | 6.43 | 6.24 | 6.18 | 5.98 | 5.97 | 5.92 |
| 190-7-30 1 | 1.02 | 0.44 | 1.67 | 2.83 | 3.96 | 4.79 | 5.55 | 5.90 | 5.91 | 5.77 | 6.01 | 5.71 | 5.65 | 5.63 | 5.66 | 5.41 |
| 190-8-30 1 | 1.02 | 0.00 | 0.42 | 1.04 | 1.94 | 2.73 | 3.64 | 4.23 | 4.49 | 4.48 | 4.78 | 4.45 | 4.50 | 4.65 | 4.59 | 4.33 |
| 190-10-30 0 | 0.00 | 0.00 | 10.6 | 10.3 | 9.88 | 9.61 | 9.33 | 9.03 | 8.77 | 8.47 | 8.18 | 7.89 | 7.63 | 7.37 | 7.09 | 6.83 |
| | | | | (k) | $p_0 = 0$ | .22 at | m; 7 | Γ ₀ = 3 | 700 K | | | | | | | |
| 190-5-3 0 | 0.60 | 0.12 | 2.52 | 3.20 | 3.17 | 3.85 | 3.95 | 3.88 | 4.03 | 3.97 | 3.99 | 4.01 | 3.82 | 3.72 | 3.72 | 3.82 |
| 1 | 0.60 | 0.00 | 1.32 | | | 3.58 | | | 3.74 | 1 | | | | 3.76 | 3.69 | |
| 190-10-3 | 0.00 | 0.00 | 6.27 | 5.98 | 5.72 | 5.46 | 5.22 | 5.05 | 4.85 | 4.68 | 4.48 | 4.34 | 4.17 | 4.03 | 3.88 | 3.75 |
| | | | (1) | p ₀ = 0 |).24 to | 0.25 | atm; | T ₀ = | 4800 | K | | | | | | |
| 190-4-9 0 | 0.60 | 0.71 | 3.37 | 4.38 | 4.48 | 5.06 | 5.38 | 5.30 | 5.54 | 5.41 | 5.50 | 5.43 | 5.23 | 5.21 | 5.19 | 5.32 |
| 190-9-6 0 | 0.60 | 0.00 | 2.47 | 3.85 | 4.62 | 5.06 | 5.54 | 5.58 | 5.51 | 5.49 | 5.57 | 5.44 | 5.42 | 5.39 | 5.30 | 4.87 |
| 190-10-6 0 | 0.00 | 0.00 | 8.62 | 8.30 | 7.95 | 7.61 | 7.32 | 7.04 | 6.77 | 6.56 | 6.31 | 6.11 | 5.90 | 5.69 | 5.49 | 5.28 |

TABLE VI.- HEAT-TRANSFER MEASUREMENTS OBTAINED FOR MODEL II - Concluded

| | Step, | Suction rate, | | H | leat-t | ransfe | r rate | e × 10 | -4, w | atts/n | 12, at | thern | nocou | ple – | | |
|---|----------|---------------|-------------------|--------------------|---------|--------|--------|------------------|--------|--------|--------|-------|-------|-------|------|------|
| Run | cm | w w | 1 | 2 | 3 | 4 | 5 | 6 | 7 | 8 | 9 | 10 | 11 | 12 | 13 | 14 |
| | , | ! | (m) | p ₀ = (| 0.36 to | 0.38 | atm; | T ₀ = | 3200 | K | L | I, | L., | L | | + |
| 190-4-12 | 0.60 | 0.61 | 1.62 | 2.30 | 2.67 | 2.68 | 2.88 | 2.88 | 3.12 | 3.14 | 3.22 | 3.11 | 3.03 | 3.07 | 2.89 | 3.19 |
| 190-5-12 | 0.60 | 0.59 | 1.68 | 2.31 | 2.51 | 2.70 | 2.86 | 2.91 | 3.34 | 3.34 | 3.35 | 3.27 | 3.19 | 3.12 | 2.97 | 3.23 |
| 190-9-12 | 0.60 | 0.00 | 1.06 | 1.74 | 2.04 | 2.32 | 2.77 | 3.13 | 2.87 | 2.87 | 3.15 | 2.75 | 2.70 | 2.91 | 2.94 | 2.81 |
| 190-10-12 | 0.00 | 0.00 | 4.73 | 4.54 | 4.38 | 4.24 | 3.98 | 3.95 | 3.82 | 3.69 | 3.54 | 3.73 | 3.27 | 3.20 | 3.12 | 3.10 |
| (n) $p_0 = 0.41 \text{ to } 0.44 \text{ atm}$; $T_0 = 3600 \text{ to } 3800 \text{ K}$ | | | | | | | | | | | | | | | | |
| 190-4-15 | 0.60 | 0.67 | 2.72 | 3.87 | 4.37 | 4.50 | 4.69 | 4.68 | 5.03 | 5.20 | 5.23 | 5.11 | 5.10 | 4.86 | 4.73 | 5.13 |
| 190-5-15 | 0.60 | 0.59 | 2.92 | 3.81 | 4.07 | 4.53 | 4.90 | 4.99 | 5.22 | 5.20 | 5.30 | 5.11 | 4.99 | 4.85 | 4.77 | 5.00 |
| 190-9-15 | 0.60 | 0.00 | 1.66 | 2.84 | 3.53 | 3.89 | 4.66 | 4.90 | 4.65 | 4.71 | 5.05 | 4.20 | 4.29 | 4.75 | 4.71 | 4.31 |
| 190-10-15 | 0.00 | 0.00 | 7.82 | 7.65 | 7.38 | 7.05 | 6.78 | 6.55 | 6.33 | 6.00 | 5.90 | 5.67 | 5.45 | 5.26 | 5.07 | 4.89 |
| (o) $p_0 = 0.64$ to 0.69 atm; $T_0 = 2900$ K | | | | | | | | | | | | | | | | |
| 190-4-21 | 0.60 | 0.70 | 1.56 | 2.20 | 2.49 | 2.58 | 2.75 | 2.81 | 2.92 | 2.96 | 3.04 | 2.63 | 2.93 | 2.84 | 2.80 | 2.91 |
| 190-5-21 | 0.60 | 0.73 | 1.78 | 2.47 | 2.70 | 2.92 | 3.05 | 3.15 | 3.31 | 3.27 | 3.31 | 3.25 | 3.22 | 3.10 | 3.03 | 3.10 |
| 190-9-21 | 0.60 | 0.00 | 0.54 | 1.15 | 1.71 | 2.06 | 2.46 | 2.65 | 2.63 | 2.67 | 2.87 | 2.62 | 2.73 | 2.70 | 2.70 | 2.54 |
| 190-10-21 | 0.00 | 0.00 | 4.53 | 4.37 | 4.17 | 4.03 | 3.87 | 3.76 | 3.64 | 3.51 | 3.40 | 3.27 | 3.16 | 3.10 | 3.01 | 2.91 |
| | • | (| p) p _C | = 0.7 | 72 to (|).77 a | tm; | $\Gamma_0 = 3$ | 400 to | 3500 | K | | | | | + |
| 190-4-24 | 0.60 | 0.73 | 2.46 | 3.58 | 3.93 | 4.07 | 4.36 | 4.45 | 4.67 | 4.74 | 4.83 | 4.71 | 4.66 | 4.41 | 4.35 | 4.54 |
| 190-5-24 | 0.60 | 0.79 | 3.43 | 4.59 | 5.20 | 5.50 | 5.69 | 5.67 | 5.84 | 5.72 | 5.79 | 5.74 | 5.61 | 5.42 | 5.38 | 5.26 |
| 190-9-24 | 0.60 | 0.00 | 0.74 | 1.68 | 1.48 | 3.12 | 3.86 | 4.22 | 4.14 | 4.18 | 4.41 | 3.88 | 4.03 | 3.63 | 4.31 | 3.85 |
| 190-10-24 | 0.00 | 0.00 | 7.25 | 6.93 | 6.68 | 6.46 | 6.24 | 6.03 | 5.84 | 5.62 | 5.42 | 5.23 | 5.09 | 4.89 | 4.68 | 4.50 |
| | | | (q) | p ₀ = 0 |).76 to | 0.82 | atm; | T ₀ = | 3600 | K | | | | | | |
| 190-4-27 | 0.60 | 0.72 | 2.94 | 4.18 | 4.70 | 4.91 | 5.21 | 5.33 | 5.58 | 5.74 | 5.69 | 5.68 | 5.66 | 5.35 | 5.21 | 5.52 |
| 190-9-27 | 0.60 | 0.00 | 0.98 | 2.12 | 3.22 | 3.92 | 4.77 | 5.10 | 5.03 | 5.13 | 5.40 | 4.86 | 4.97 | 5.20 | 5.17 | 4.76 |
| 190-10-27 | 0.00 | 0:00 | 8.81 | 8.51 | 8.21 | 7.91 | 7.61 | 7.37 | 7.09 | 7.83 | 6.56 | 6.32 | 6.10 | 5.83 | 5.62 | 5.40 |
| | + | , | (r) | p ₀ = 0 | 0.82 to | 0.87 | atm; | T ₀ = | 3900 | K | | | | اسبب | _ | |
| 190-4-30 | 0.60 | 0.78 | 3.71 | 5.39 | 6.02 | 6.28 | 6.68 | 6.80 | 6.75 | 6.79 | 7.11 | 7.21 | 7.29 | 7.14 | 7.05 | 6.78 |
| 190-9-30 | 0.60 | 0.00 | 0.64 | 2.79 | 4.10 | 5.01 | 6.07 | 6.31 | 6.32 | 6.47 | 6.63 | 6.11 | 6.28 | 6.49 | 6.48 | 5.88 |
| 190-10-30 | 0.00 | 0.00 | 10.5 | 10.2 | 9.85 | 9.56 | 9.26 | 8.95 | 8.65 | 8.41 | 8.12 | 7.87 | 7.60 | 7.11 | 7.07 | 6.79 |

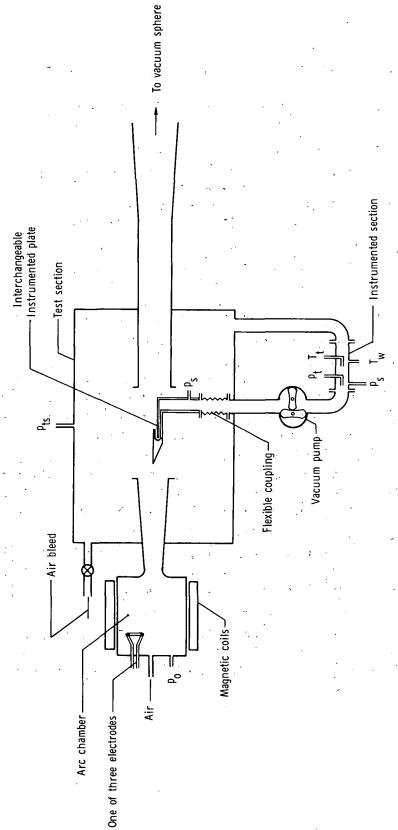


Figure 1.- Schematic diagram of suction experimental setup.

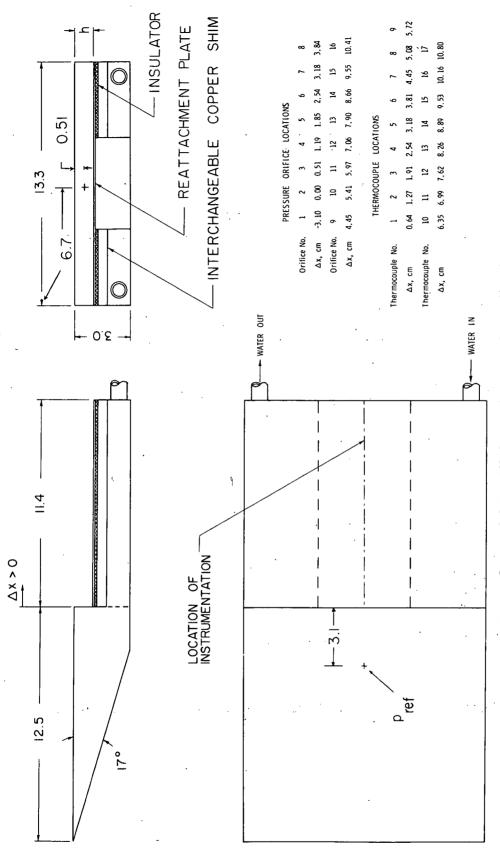


Figure 2.- Model I. All dimensions are in centimeters.

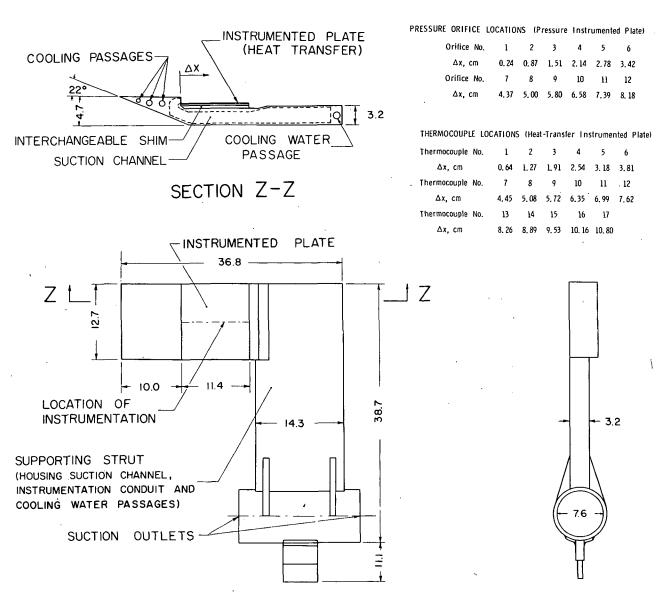


Figure 3.- Model II. All linear dimensions are in centimeters.

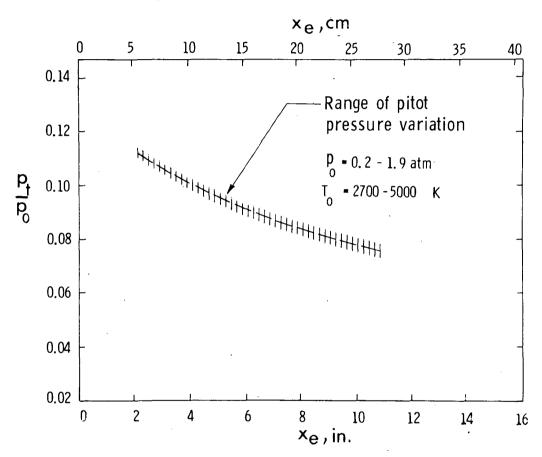


Figure 4.- Axial distribution of pitot pressure in test section.

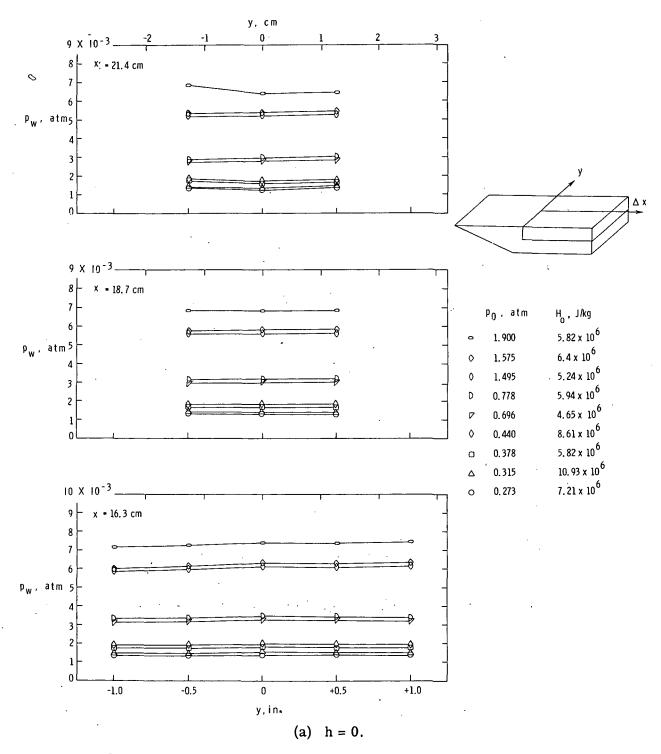


Figure 5.- Transverse surface pressure distribution on model I.

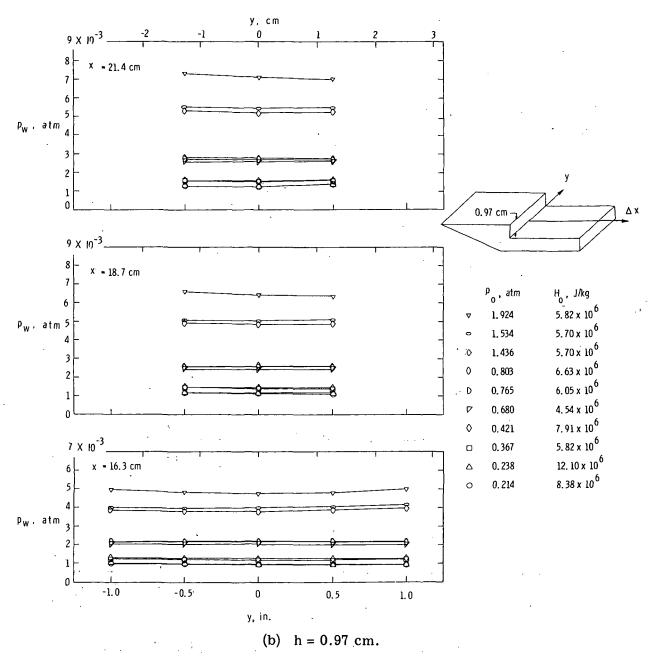


Figure 5.- Concluded.

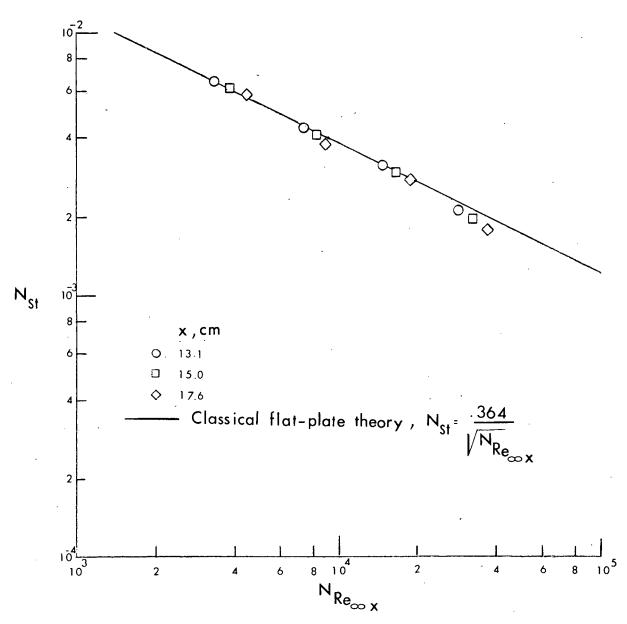


Figure 6.- Stanton number correlation of flat-plate heat-transfer data as a function of Reynolds number.

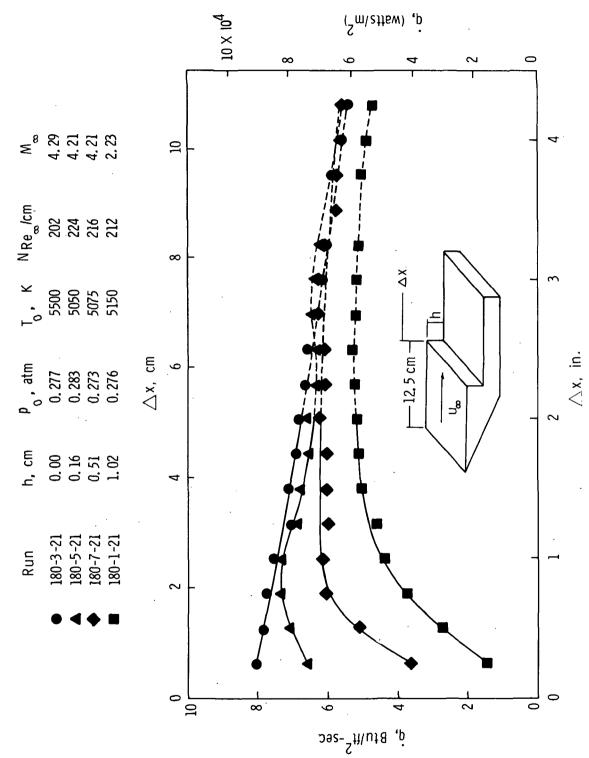


Figure 7.- Effect of step height on heat-transfer distribution downstream of step. Model I (no suction).

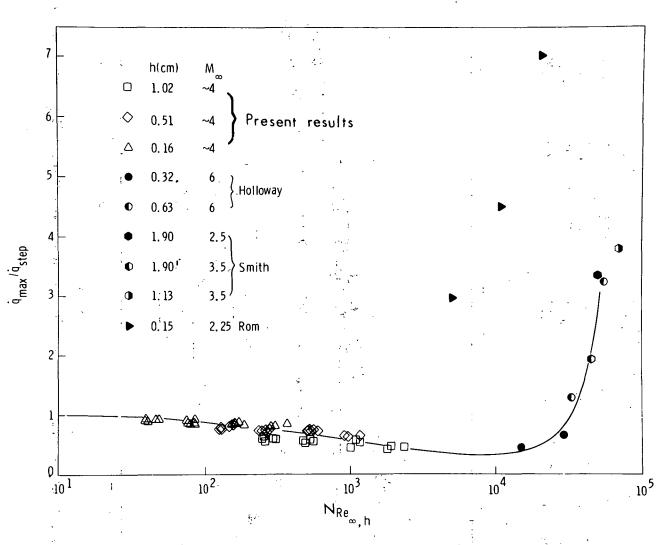


Figure 8.- Maximum heating rate as function of $N_{\mathrm{Re}_{\infty}}$ h

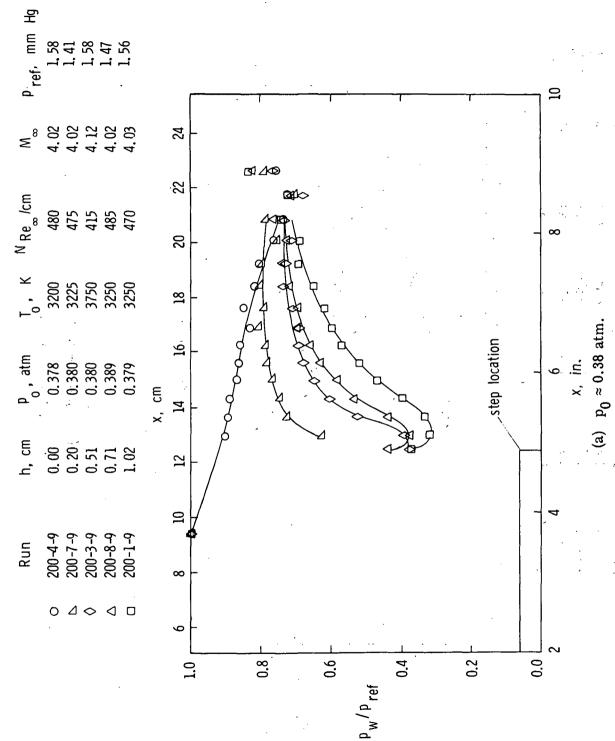
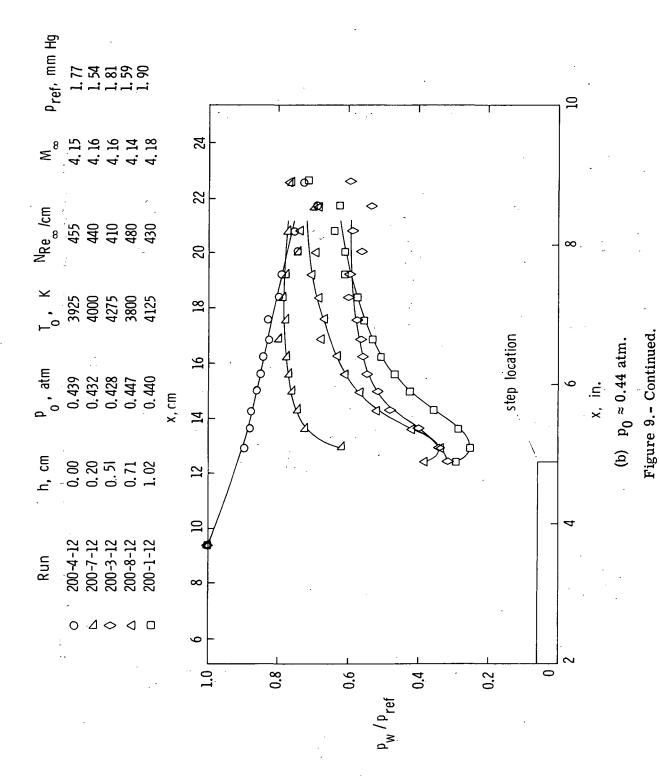
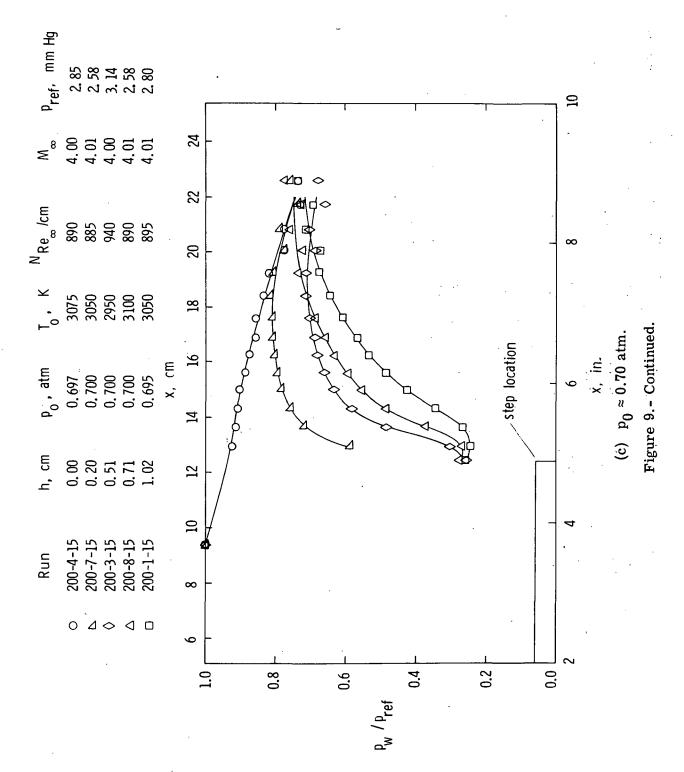
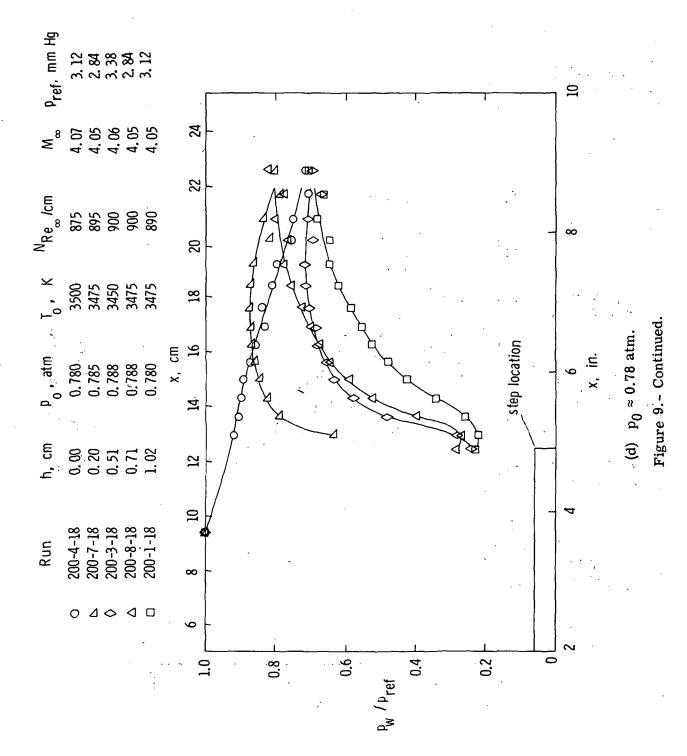
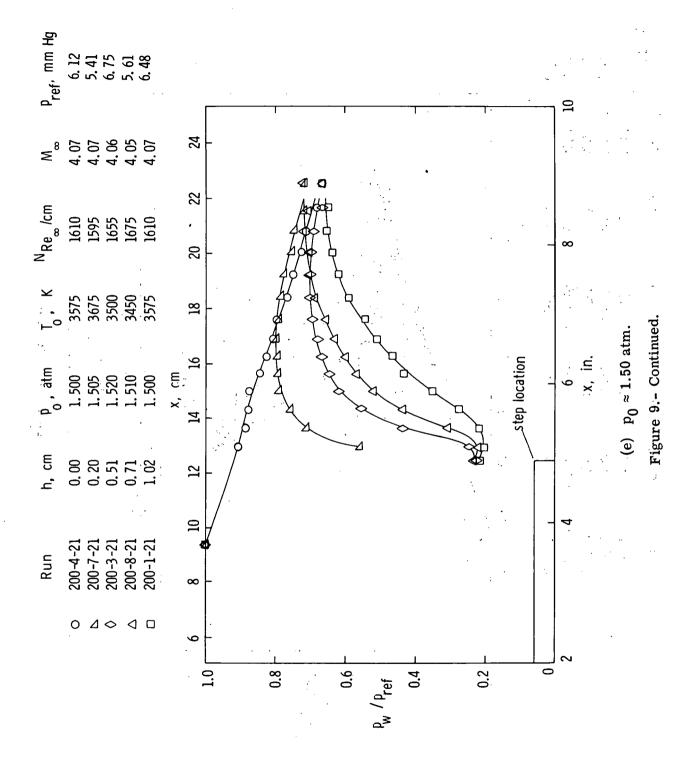


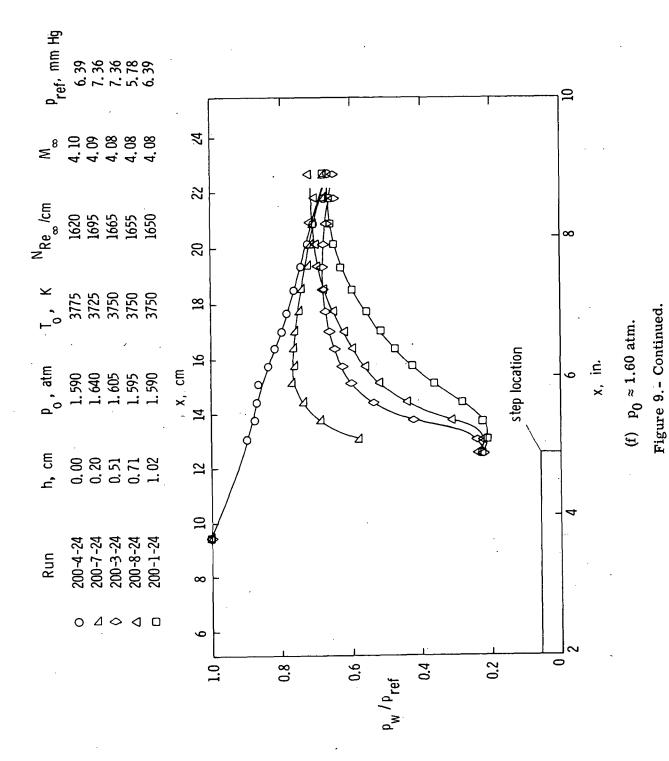
Figure 9.- Normalized wall pressure in the separated region on model 1

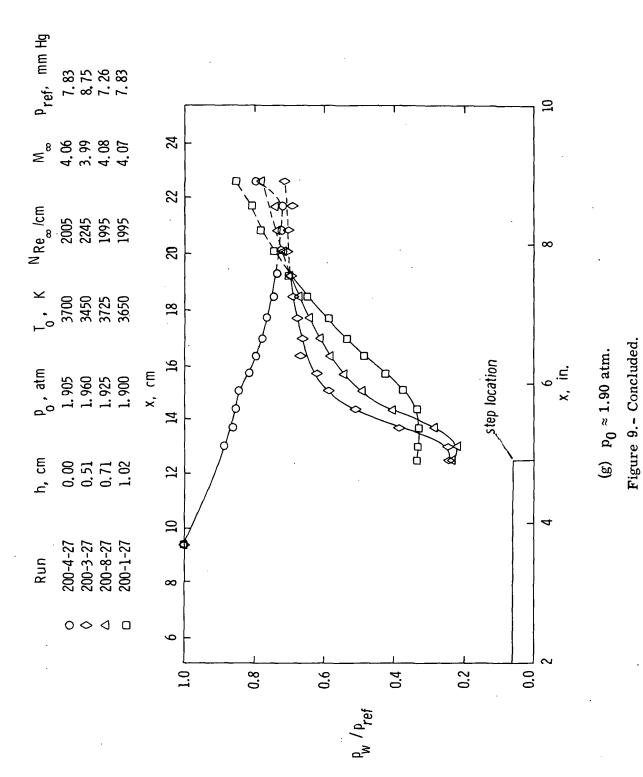












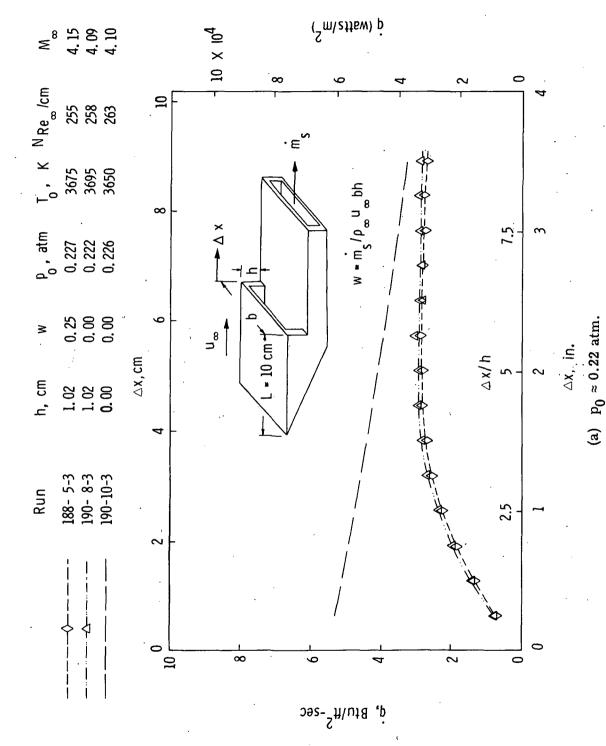
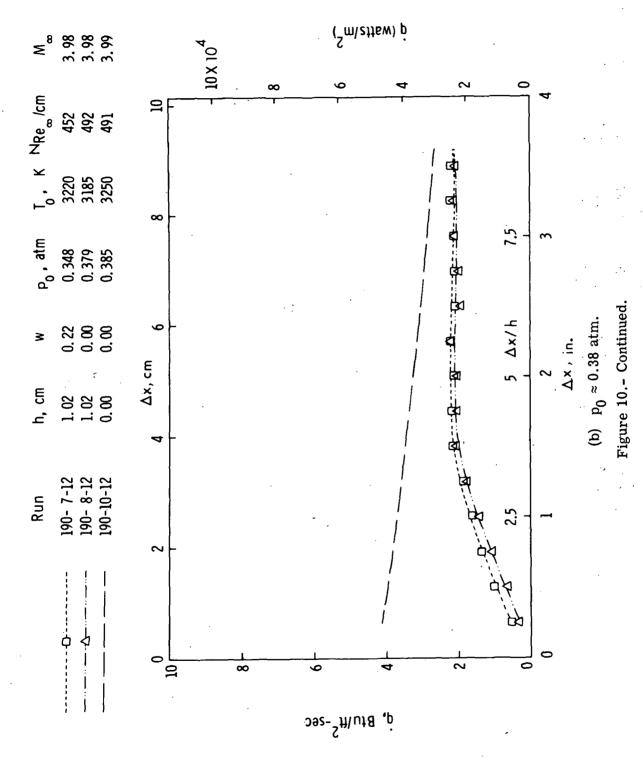
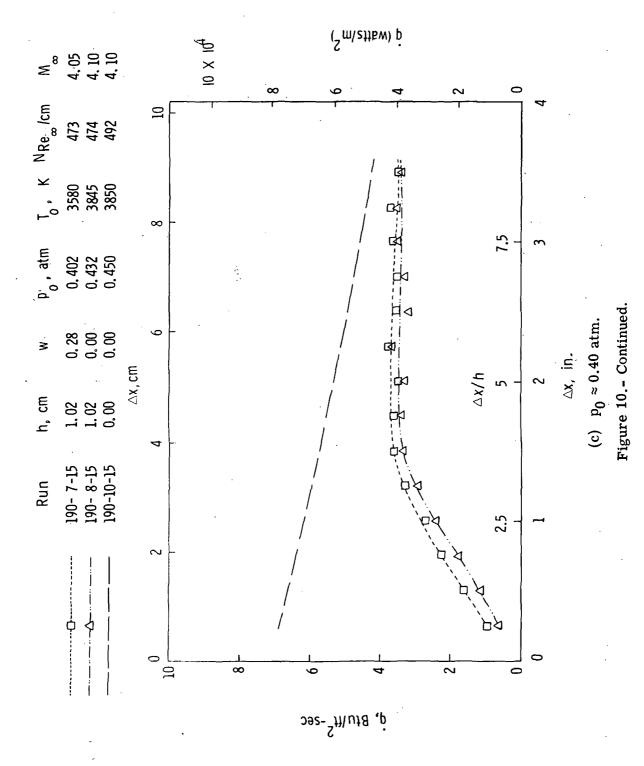
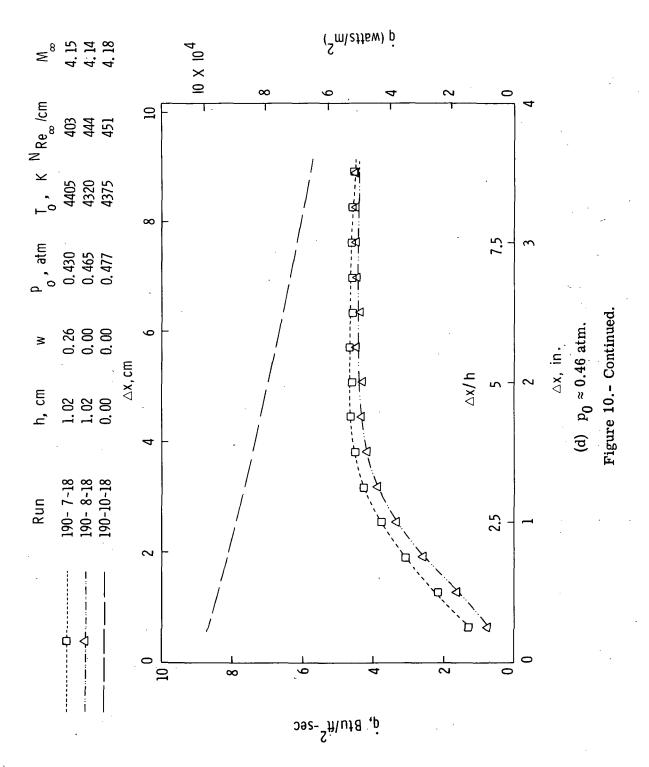
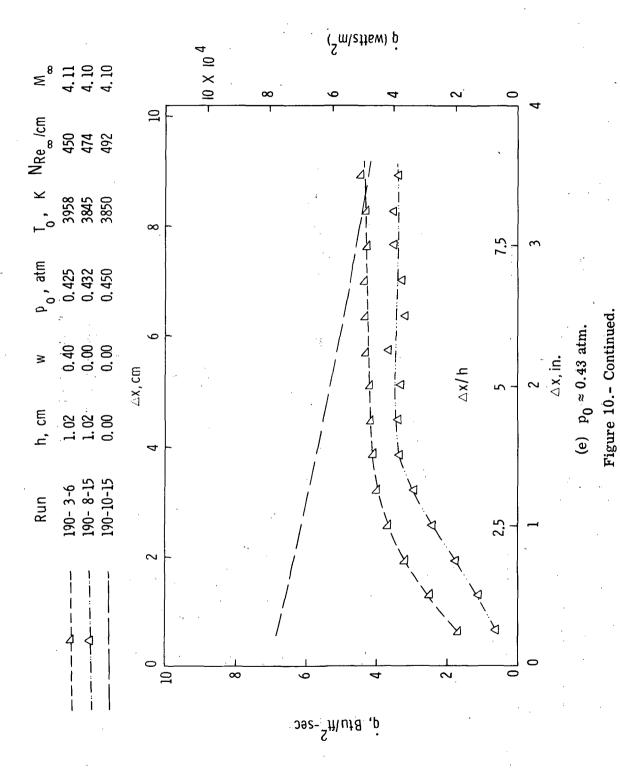


Figure 10.- Effect of mass suction on heat-transfer distribution downstream of a 1.02 cm step.









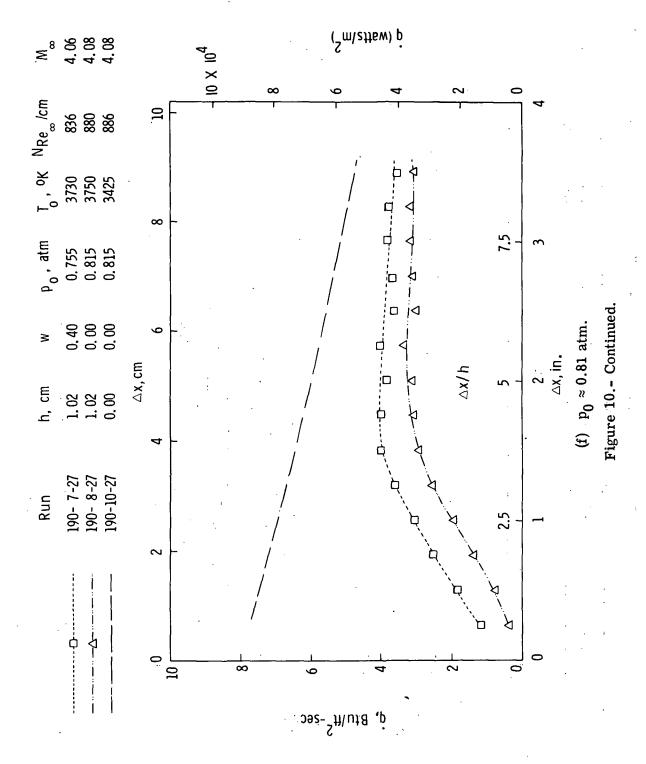
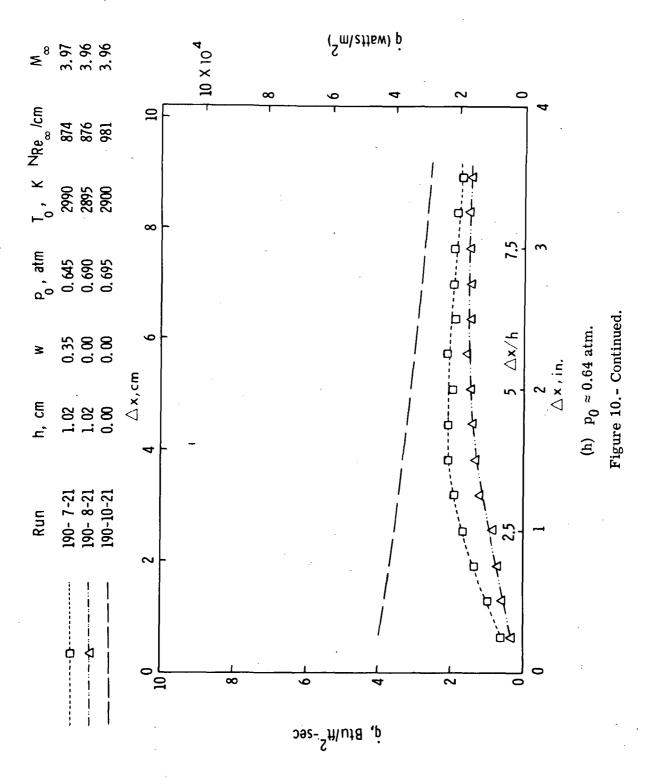
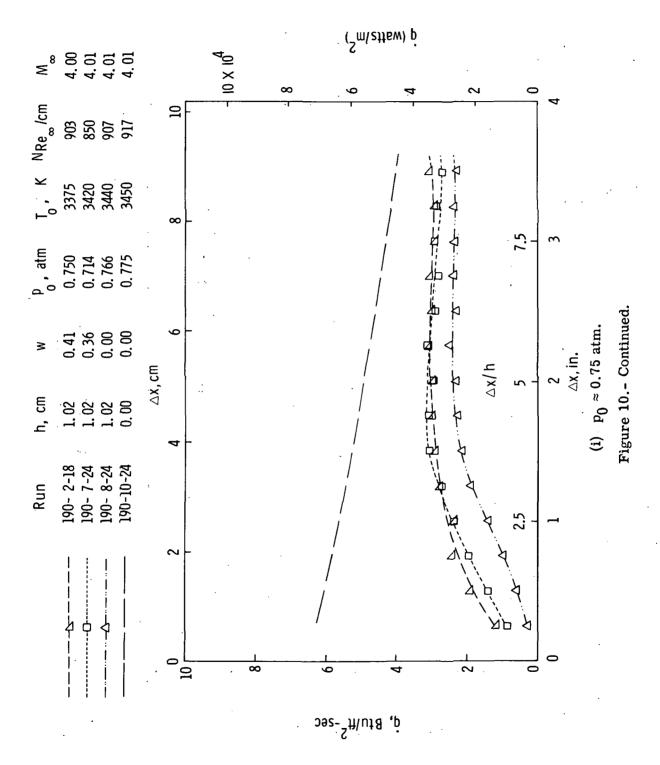
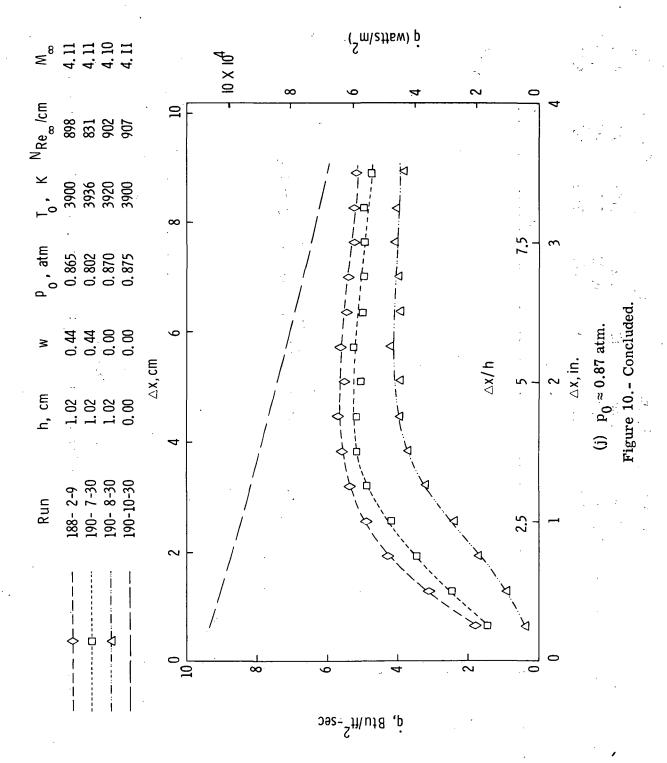


Figure 10.- Continued.







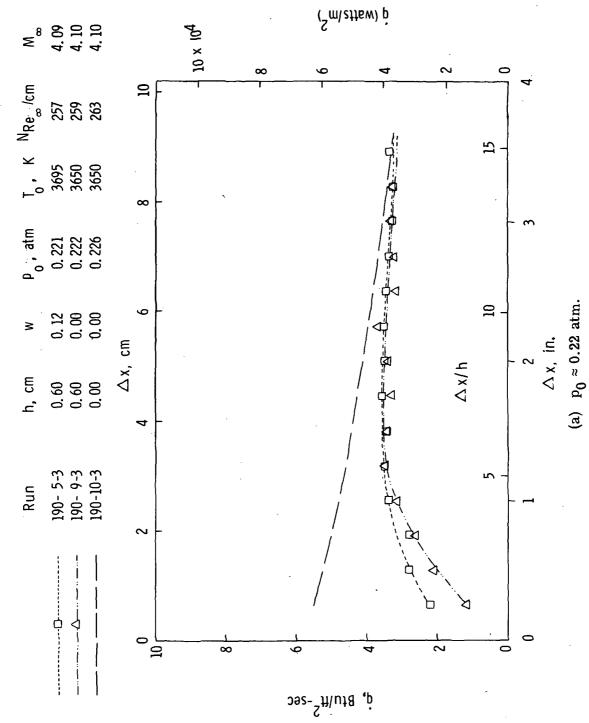
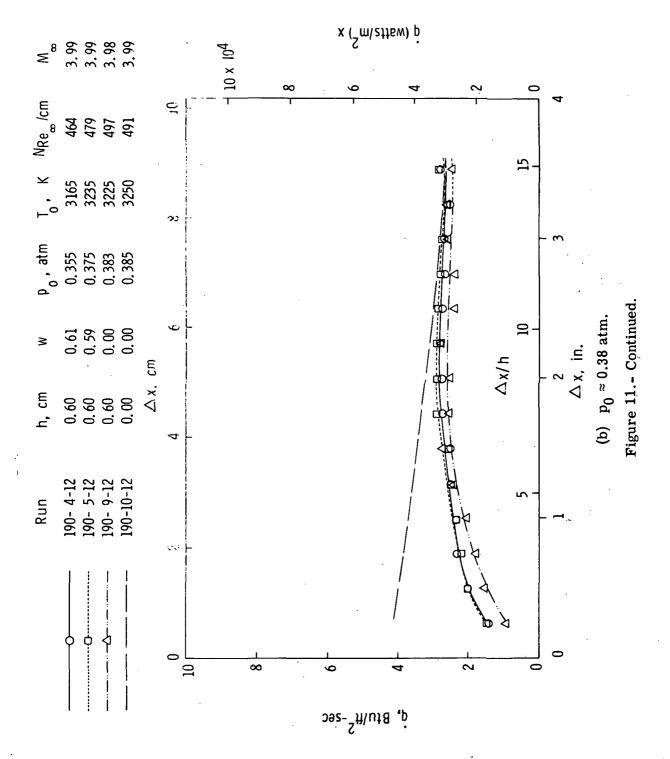
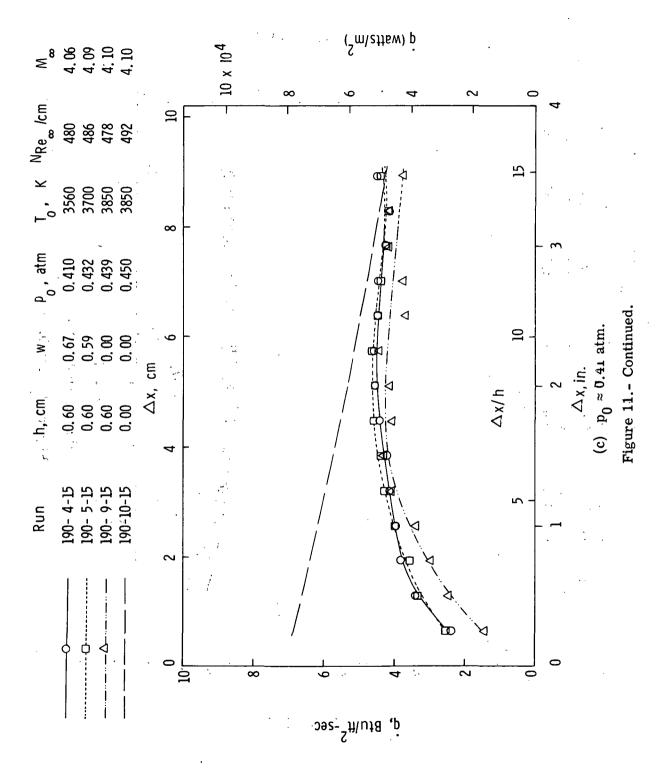


Figure 11.- Effect of mass suction on heat-transfer distribution downstream of a 0.6 cm step.





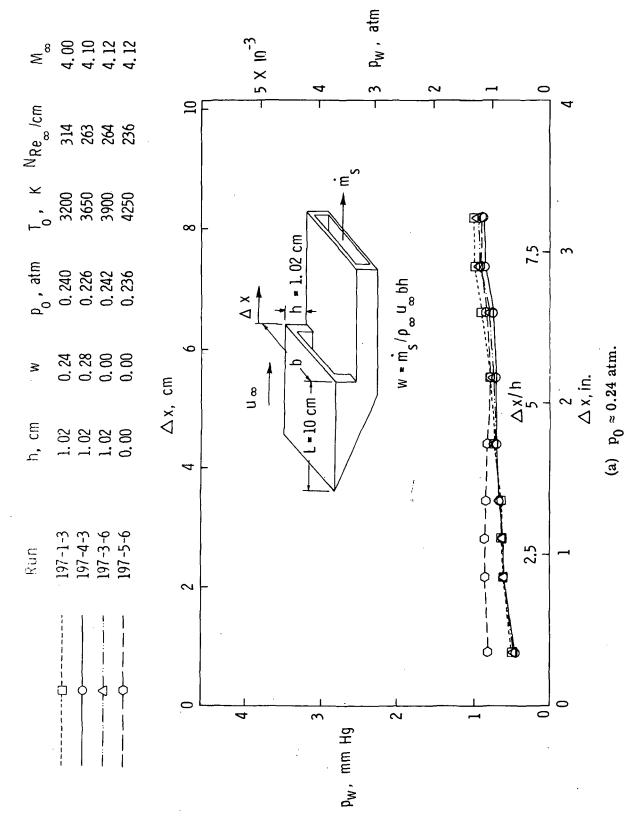


Figure 12.- Effect of mass suction on pressure distribution downstream of a 1.02 cm step.

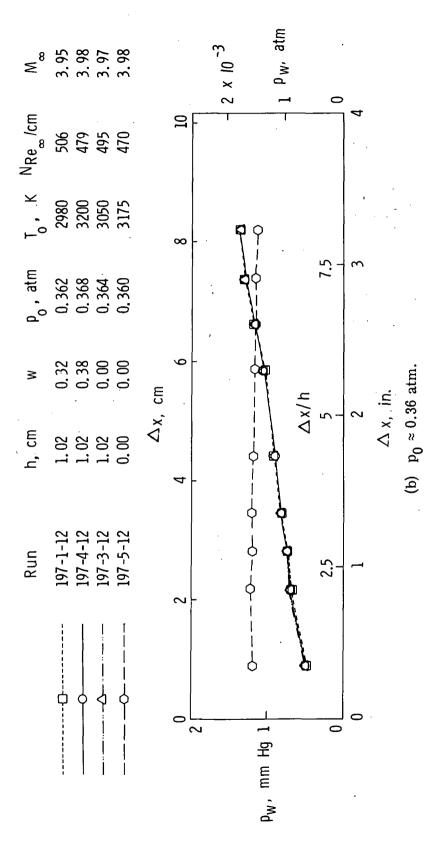
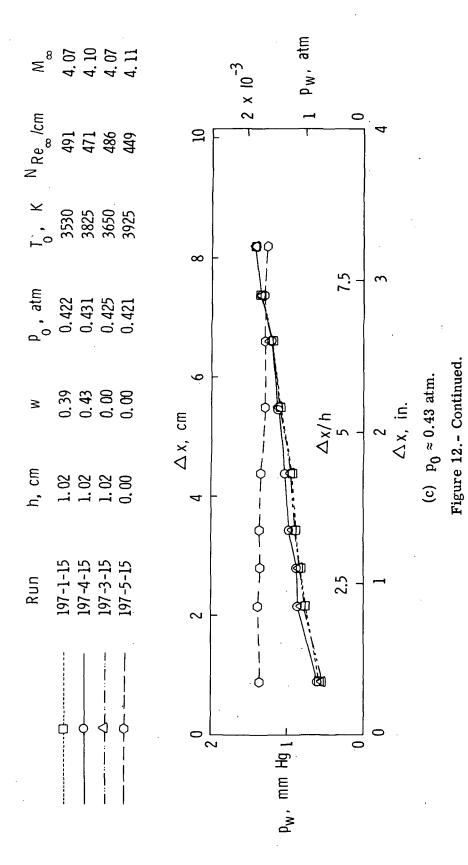
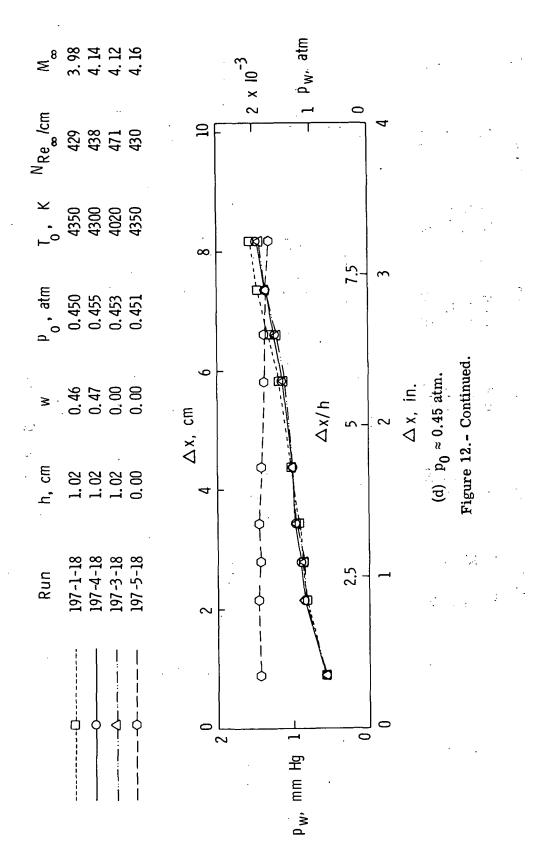
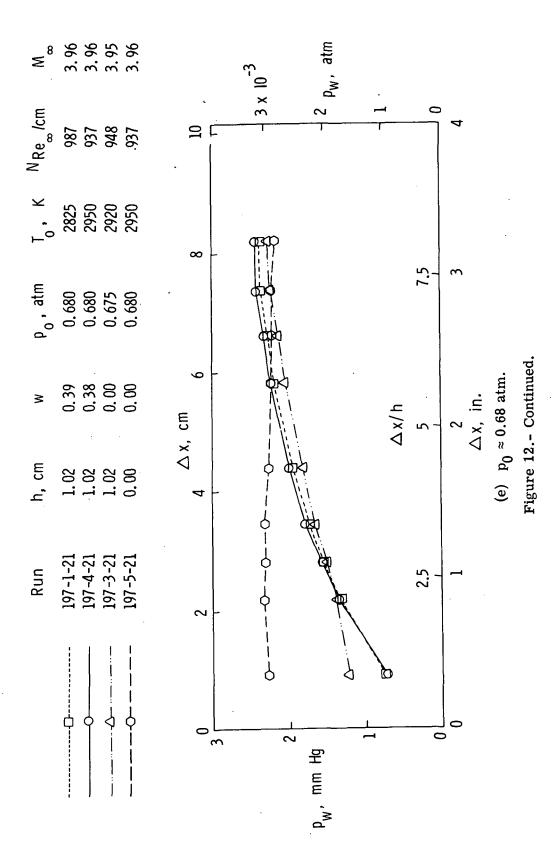
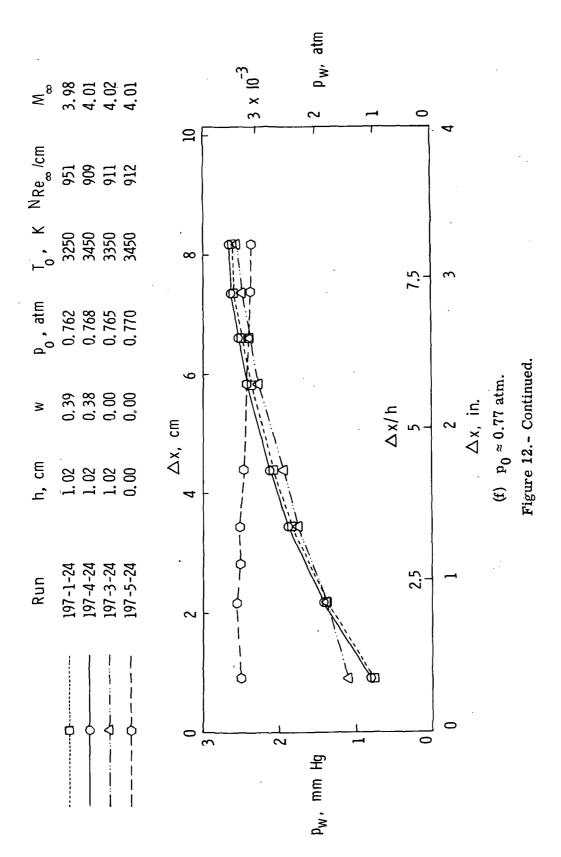


Figure 12.- Continued.









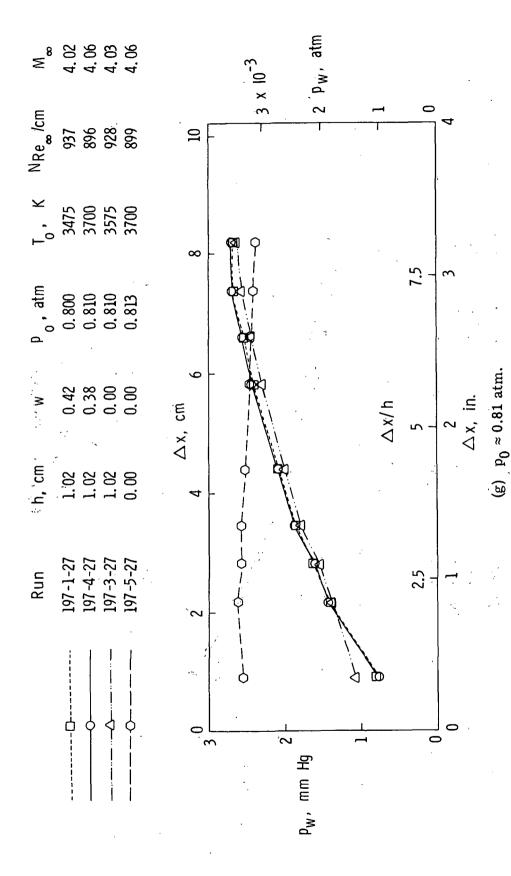
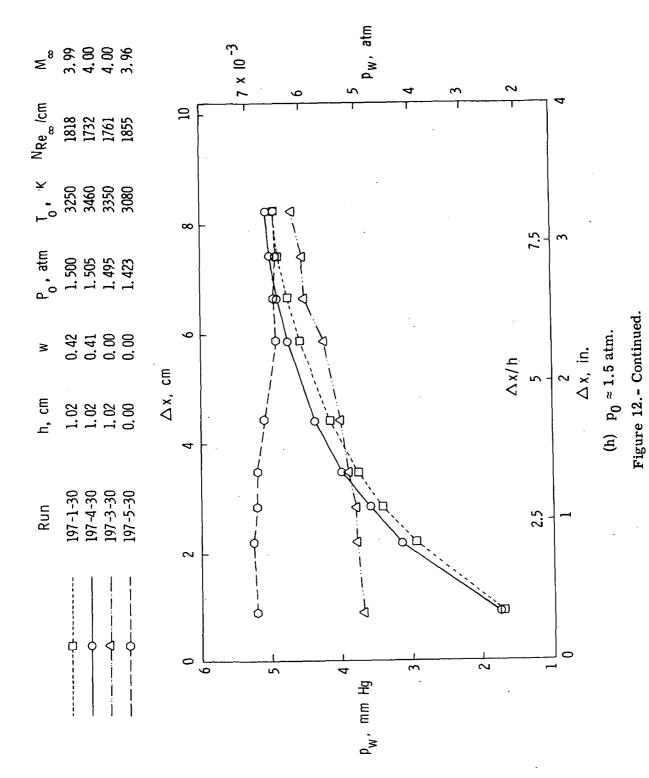


Figure 12.- Continued.



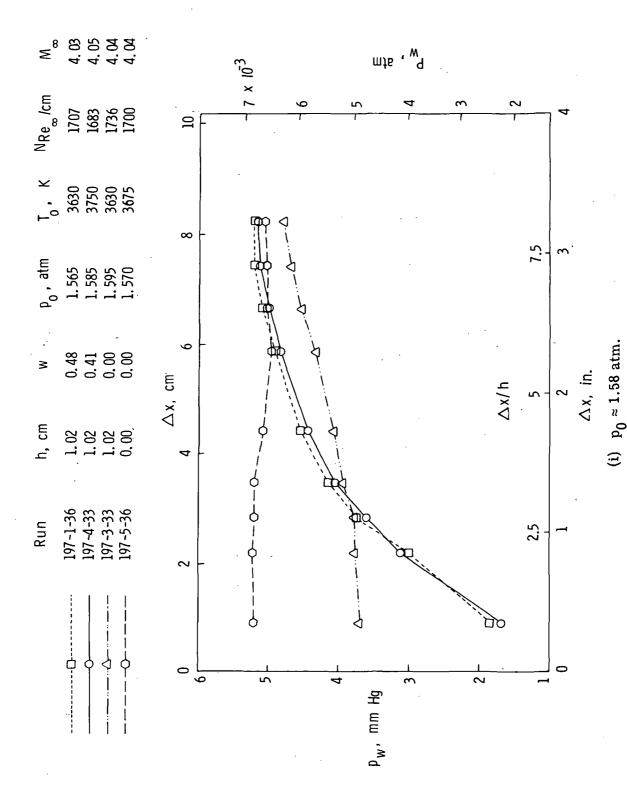
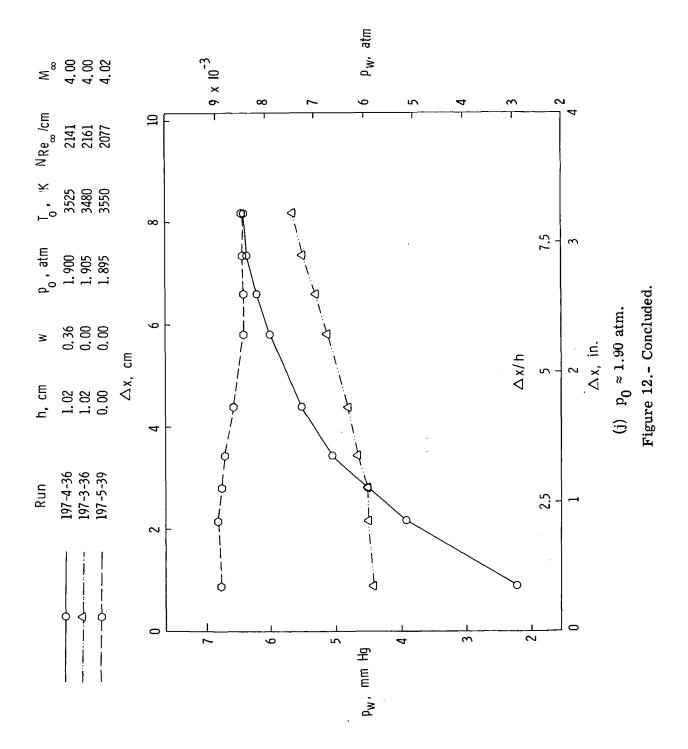


Figure 12.- Continued.

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